

Lasers for Ranging and Satellite Tracking

A brief technology overview.

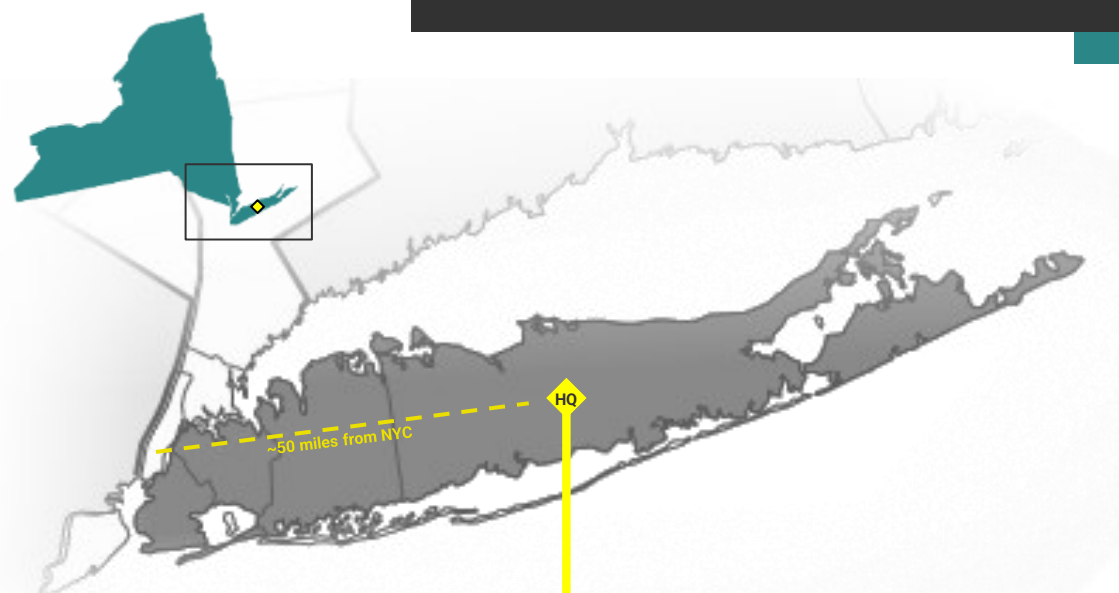
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Hans Taylor – MD – Photonics Industries BV – Netherlands.

A solid teal horizontal bar spanning the width of the slide at the bottom.

Contents

- ❖ Who are we?
- ❖ Low Frequency – High Pulse Energy – Picosecond Lasers
 - ❖ Mostly Cooperative targets
- ❖ High Frequency – High Pulse Energy – Picosecond Lasers
 - ❖ Cooperative targets
- ❖ Low Frequency – Very High Pulse Energy – Nanosecond
 - ❖ Uncooperative targets.
- ❖ Emerging Technologies - Mid PRF – Tuneable USP Pulse



Photograph illustration view of New York City from Jersey City

PI AT A GLANCE



“The pioneer of intracavity harmonic lasers”
Manufacturer of ns, ps, sub-ns, and fs lasers



Founded in 1993
Celebrating 31 years a company



International, worldwide company
>250 employees worldwide



Headquarters and manufacturing
Ronkonkoma, New York, USA



Privately owned and operated



Vertically Integrated Factory
6750 sqm, 6.5 acre lot
4000 sqm of clean room



Technology Leader
30+ Patents Granted or Pending

NANO/PICO/FEMTOSECOND LASERS

Photonics Industries International, Inc. (www.photonix.com) is the pioneer of intracavity harmonic lasers, and manufactures a wide range of lasers in the **nanosecond**, **sub-nanosecond**, **picosecond**, and **femtosecond** regimes



Intracavity Ultraviolet and Green Q-switched nanosecond lasers

Short pulse ~ 4 to 50ns Intermediate ~ 80ns and long pulse >300ns

From **500mW** to **50W UV** and Green to **>400W**

High pulse energies from μ J-level to **>200mJ**

Pulsed Pumped Q-switched nanosecond lasers

Short pulse 8 ns and up to 1000Hz

TEM₀₀: **20 to 50mJ** & Multimode: **>200mJ**

Harmonics Wavelengths: Green | UV | Deep UV



Picosecond and Sub-Nanosecond lasers

Picosecond - **10 to 7ps** - Sub-Nanosecond – 100ps to 5ns

From **10W** to **>150W IR**

Harmonics Wavelengths: Green | UV | Deep UV

Femtosecond lasers

Ultrashort pulses down to ~450fs

From **5W** to **100W IR**

Harmonics Wavelengths: Green | UV | Deep UV



Dedicated to OEM Laser Manufacture

- ❖ Only Make Laser Sources
- ❖ Only make solid-state DPSS Lasers
- ❖ Six product ranges – 100+ individual models
- ❖ 95% of our output goes directly to industrial systems
- ❖ Our installed based measures >40,000 units.

You will today, use, wear, watch or interact with a piece of technology that was directly or in some part made or enabled by using a PI laser.

LOW PRF – HIGH PULSE ENERGY LASERS

RGLX Series – was designed specifically to meet the specifications provided by NASA.

Wavelength: 1064nm or 532nm
Pulse Duration: <30 ps
Pulse Frequency: 1000Hz – Master 1 to 5 kHz capable.
Pulse Energy: 2 mJ or 4 mJ @ 1064nm
1.5 mJ or 2.5 mJ @ 532nm
Beam Mode: True TEM₀₀ M2 < 1.3 (ISO)
Pointing Stability: < 50 μrad – full angle (ISO)

Exceptionally Compact Form Factor.
Optionally: air-cooled.

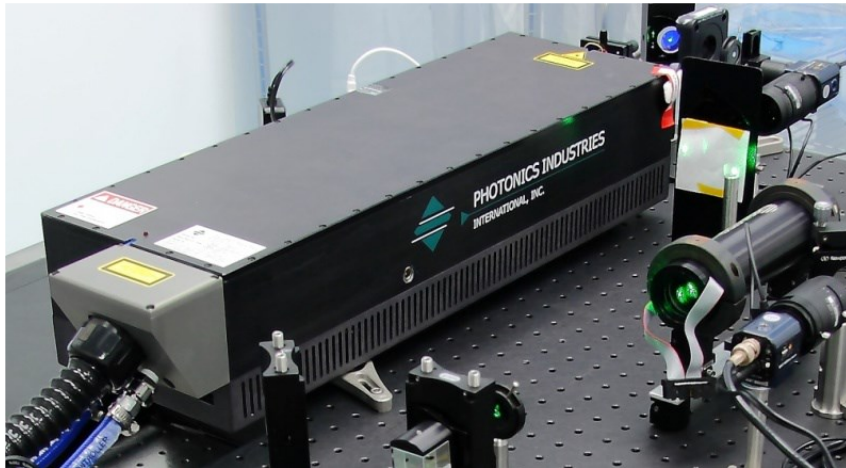
Technology Overview:

Mode-Locked Master Oscillator
Free Space Regenerative Amplifier – Diode (Fiber pumped)
Pockels Cell Based Pulse Picker

532nm – “Extra cavity” – Using a RX based Optical Sub Assembly.

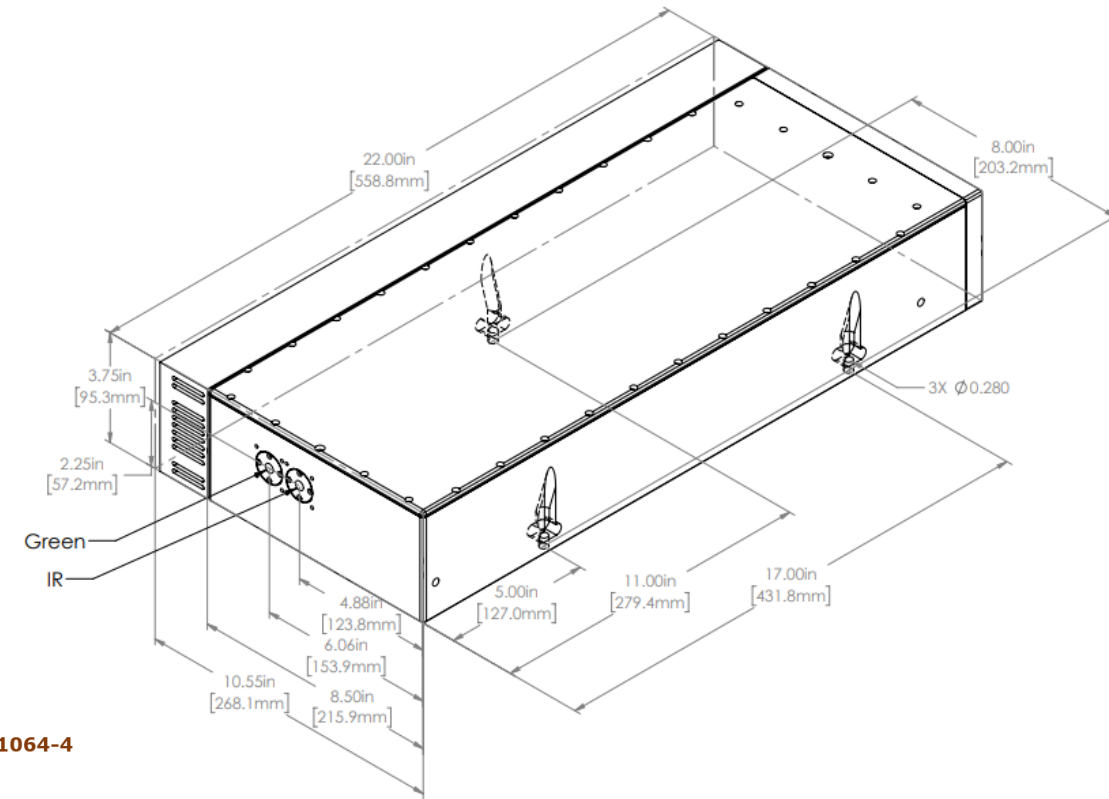
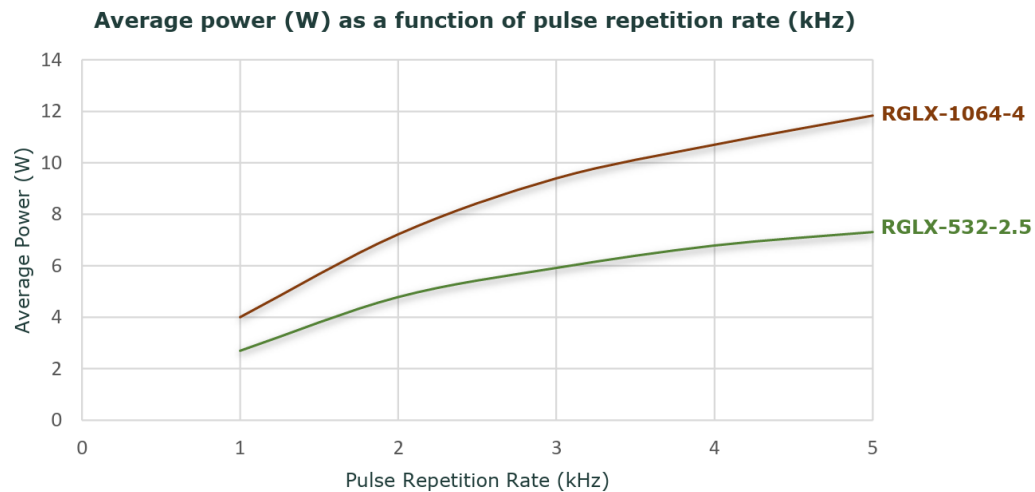


LOW PRF – HIGH PULSE ENERGY LASERS



The first RGLX Image: Courtesy NGSLR – Greenbelt

~ 50ps - 2.7 mJ @ 532.19nm



Current RGLX – Air Cooled – All In One Laser
No External Water or 19" Rack PSU.

HIGH PRF – HIGH PULSE ENERGY LASERS

RX-HE Series – Developed for industrial hard materials processing.

Wavelength: 1064nm
Pulse Duration: 10 ps
Pulse Frequency: 100,000Hz – Master
PRF Range: 1kHz to > 1MHZ
Pulse Energy: 650 μ J @ 100KHz
 900 μ J @ 100KHz
Beam Mode: True TEM₀₀ M2 < 1.3 (ISO)
Pointing Stability: < 25 μ rad – full angle (ISO)

Exceptionally Compact Form Factor.

Technology Overview:

Mode-Locked 100% Fiber Seeder and Pre-amplification
 Free Space – Multi-stage Amplifier – Diode (Fiber pumped)
 Acousto-Optic Modulator Based Pulse Picker
 AOM – Pulse Energy Control – Attenuator. 100% to >1%

532nm – “Extra cavity” – Using a RX based Optical Sub Assembly.

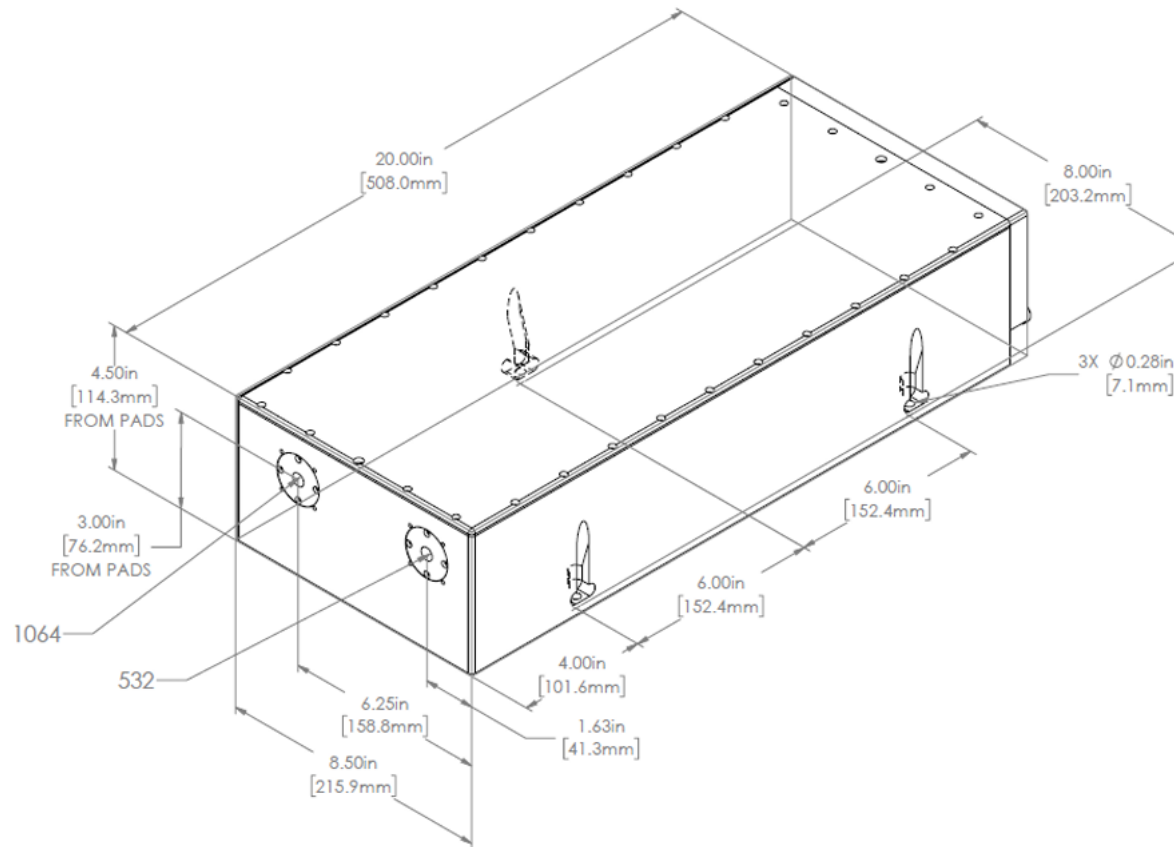


Exceptionally Compact Form Factor.

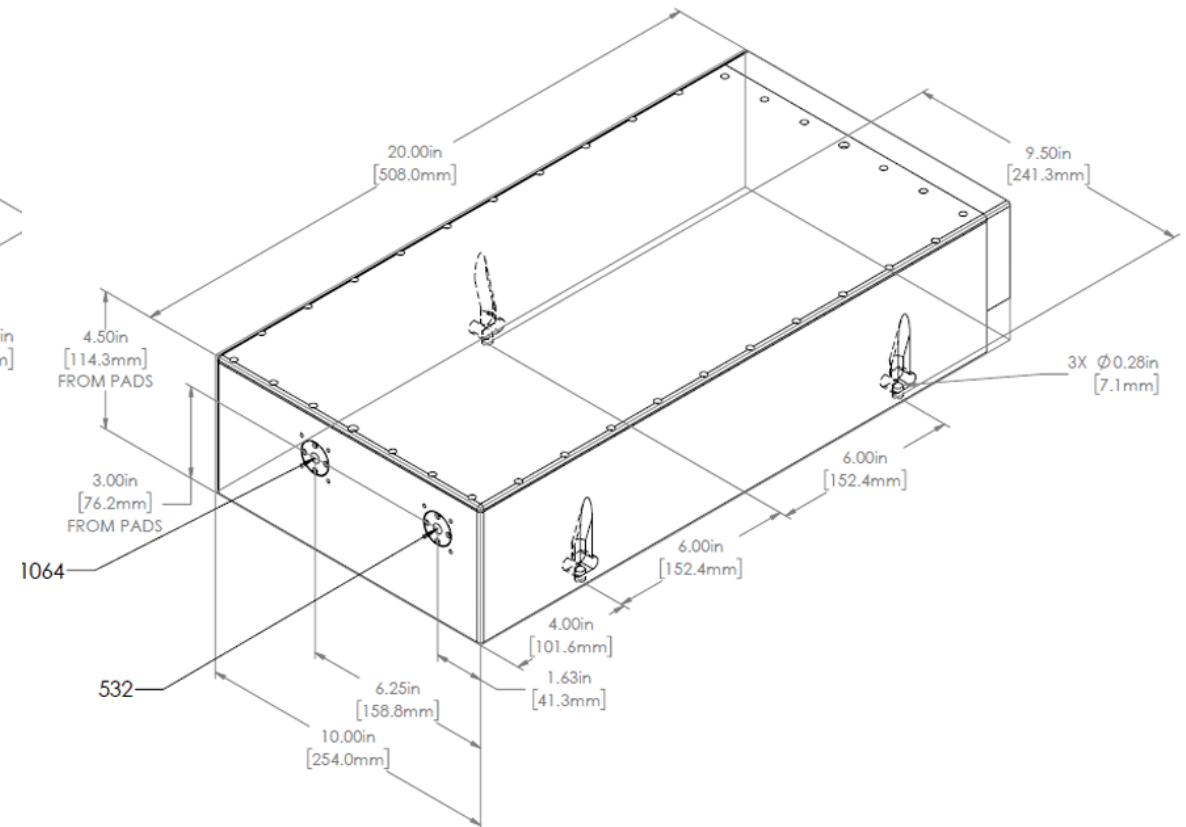
HIGH PRF – HIGH PULSE ENERGY LASERS

RX-HE Series – Form Factors

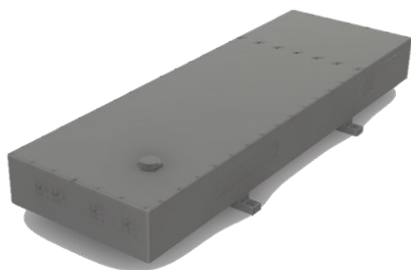
RX-100-HE - 508 x 115 x 203mm – 21.5 Kg
65W - 650 μ J @ 100 kHz



RX-150-HE - 508 x 115 x 241mm – 26 Kg
95W – 950 μ J @ 100 kHz



RX Series Journey - Continuous Innovation

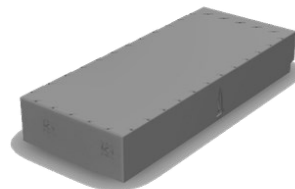


2015

No separate controller, All-in-one (AIO) ps Laser

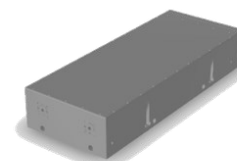


2018/19



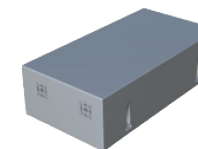
2020

35% Smaller form factor from predecessor ps laser



2022

25% Smaller form factor from predecessor ps laser



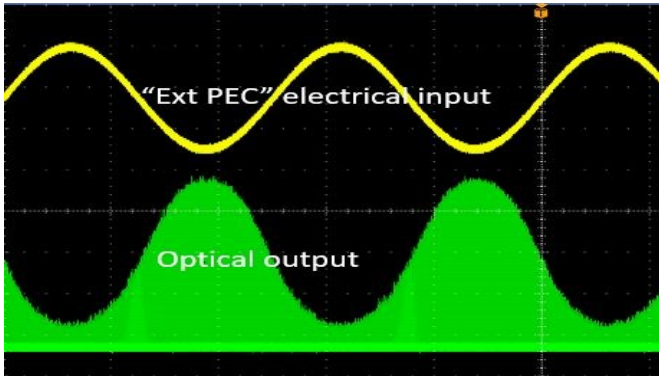
2025

~10% Smaller form factor from predecessor ps laser

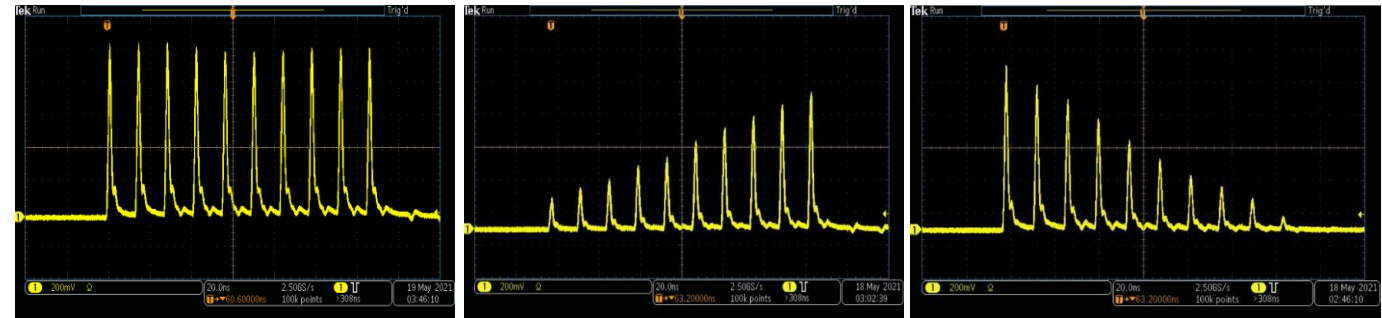
- ❖ **Flex-Pulse™** offered by the RX Series Lasers provide complete control over the pulsed output:
 - ❖ **PEC** (Power or Pulse Energy Control) controlling the overall optical energy from the laser
 - ❖ **PSO** (Position Synchronized Output) support for external triggering to any arbitrary PRF while maintaining a constant, stable pulse energy with low jitter
 - ❖ **Digi-Burst™** for individually controllable bursts of up to 10 pulses with a separation of about 31 ns
 - ❖ **POD** (Pulse-On-Demand) pulse bursts can be triggered internally, externally, or continuously, while maintaining constant pulse energy

RX Series – FLEX-Pulse™

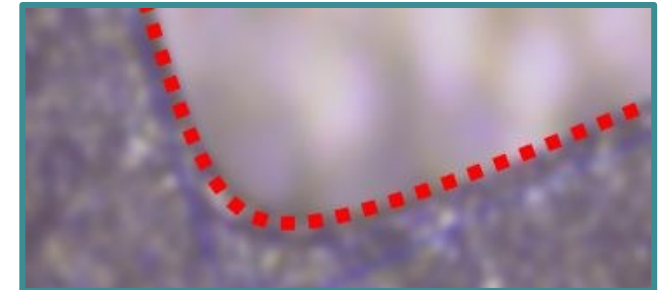
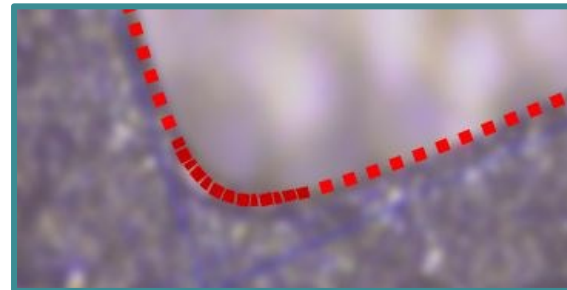
- ❖ **PEC** (Power or Pulse Energy Control) controlling the overall optical energy from the laser



- ❖ **Digi-Burst™** for individually controllable bursts of up to 10 pulses with a separation of 31 ns



- ❖ **PSO** (Position Synchronized Output) support for external triggering to any arbitrary PRF while maintaining a constant, stable pulse energy with low jitter.

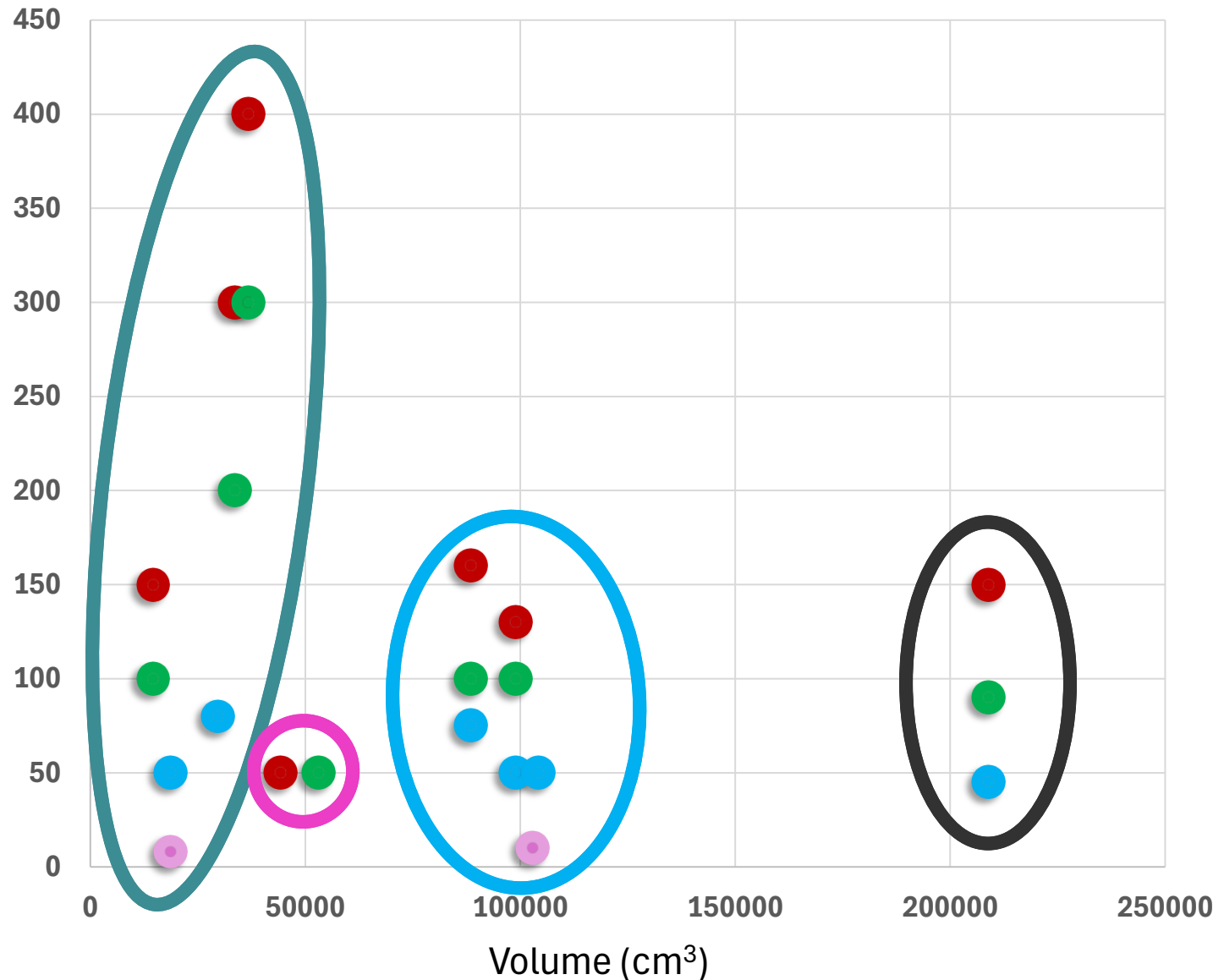


RX-GEN 4

- ❖ Hybrid Laser Configuration
 - ❖ Seeder – 100% fiber
 - ❖ OSA's already very small
 - ❖ Onboard Electronics (water/cooled)
 - ❖ Integrated 2HG if specified.
 - ❖ All DC and highly efficient
 - ❖ Fully temperature stabilised optical base
- ❖ Free Space Amplification:
 - ❖ All w/c solid state amplifiers
 - ❖ Multi-stage using proven DPSS Technology
 - ❖ Flexible power scaling - Simple/reliable
 - ❖ Pulse Energy biased
- ❖ Size to power comparison:
 - ❖ Green – PI
 - ❖ Pink – MKS
 - ❖ Blue – Lumentum / Coherent
 - ❖ Black - Trumpf

Power (W)

Ps Power vs Volume



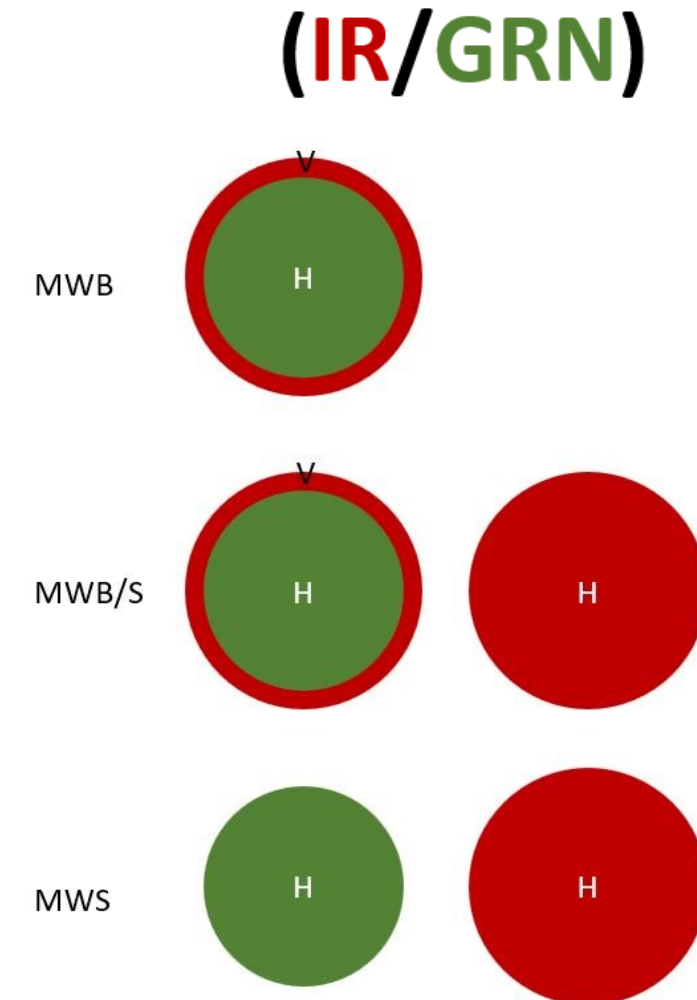
RX Series – Wavelength Switching Features

MWB – Multiwavelength Blended

Both wavelengths come out of a single exit port all blended together.

MWS – Multiwavelength Switchable

IR or green emitted from the same single exit port. The user selects which individual single wavelength is emitted at any given time via the GUI. The advantage is having access to all the available wavelengths collinearly from a single exit port with fast (<3 sec) wavelength switching on the fly.



LOW PRF – VERY HIGH PULSE ENERGY LASERS

❖ Traditionally Lamp-Pumped Q-Switched Lasers

- ❖ > 1 to 5J @ 532nm required
- ❖ Very large | Very Heavy Optical Heads
- ❖ Very Large 19" Rack Mounted PSU
- ❖ <1 % Wall plug to optical efficiency
- ❖ Very Larger ~ 4kH water cooling
- ❖ Large Format – very expensive 2HG Crystal
- ❖ HV Lamp PSU and frequent lamp changes.

Lamp PSS - 2100 mm x 210 x 330mm – 150 Kg

❖ 2025 – DPSS

- ❖ New DPSS Pumping Chambers
- ❖ Over Design Concept Unchanged
- ❖ Large Frame Optical heads.
- ❖ Optical Tables and Heavy Mechanical optical mounts.

DPSS - 1300 mm x 210 x 330mm – 100 Kg

LOW PRF – VERY HIGH PULSE ENERGY LASERS

DPM Series – Developed for industrial semiconductor processing

Wavelength: 1064nm or 532nm
Pulse Duration: ~ 10ns
Pulse Frequency: 10Hz
Pulse Energy: 4 J @ 1064 nm
2 J @ 532 nm
Beam Mode: Multi-mode : M2 < 8 (ISO)
Pointing Stability: < 50 µrad – full angle (ISO)

Exceptionally Compact Form Factor.

Water Cooled

**Onboard: Mode Monitoring / 100% Energy Monitoring /
532 nm Auto-Tuning.**

Technology Overview:

Compact Low M2 Master Oscillator

Free Space Amplifiers – Direct Diode Pumped – Patented Designs

Pockels Cell Q-Switch

532nm – “Extra-cavity” full auto-tune – 100% beam dump shutter.

QI – Quite Interesting



MID PRF – HIGH PULSE ENERGY LASERS

SN Series – Developed for industrial hard materials processing.

Wavelength: 1064nm & 532nm
Pulse Duration: 100 ps to 5 ns - Selectable
Pulse Frequency: 10kHz
 100,000Hz
Pulse Energy: 4+ mJ @ 10KHz (2.5mJ @ 532nm)
 2.5 mJ @ 100KHz (1.5mJ @ 532nm)
Beam Mode: True TEM₀₀ M2 < 1.3 (ISO)
Pointing Stability: < 25 μ rad – full angle (ISO)

Exceptionally Compact Form Factor.

Technology Overview:

Seeder: Proprietary DPSS USP Seeder – 100ps to 5ns
 Free Space – Multi-stage Amplifier – Diode (Fiber pumped – RX Series)
 Acousto-Optic Modulator-Based Pulse Picker
 AOM – Pulse Energy Control – Attenuator. 100% to >1%

532nm – “Extra cavity” – Using a RX based Optical Sub Assembly.



Same form factor as RX Series



Thank you for your time.

Any Questions?



Omni-SLR in 2025: What's new and what's next

This work has been supported by JSPS KAKENHI Grant Number 20H01993 and 25K01089 , and the joint research projects of JAXA-HitU-NAOJ-UTokyo, GSI-HitU, NICT-HitU, NIPR Exploratory Research, Softbank-HitU, MHI-HitU and KHI-HitU.

Toshimichi Otsubo [1], Hiroshi Araki [2], Yusuke Yokota [3], Kenji Kouno [3],
Mihoko Kobayashi [1], Akihisa Hattori [4], Yuichi Aoyama [4],
Haruna Furui [5], Shino Araki [5]
and a number of contributors/supporters

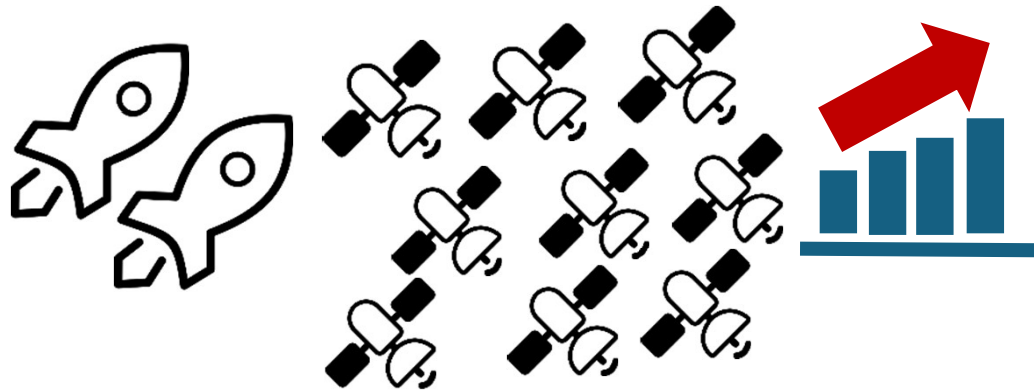
[1] Hitotsubashi Univ, [2] NAOJ, [3] IIS, Univ of Tokyo, [4] NIPR, [5] GSI



Omni-SLR: new attempts for SLR



SLR Targets: monotonically increasing



Very Compact

High mobility < 100 kg. Low energy consumption < 100 W.

Very Low-cost

A bit more expensive in 2025

Hardware: net 50-60k EUR/USD. Key components = COTS.

Multi-purpose

Primary: SLR.

Linked with various applications

(time transfer, space comm, 6G airborne comm, etc).

SLR Stations: almost the same

Current and Planned ILRS Network

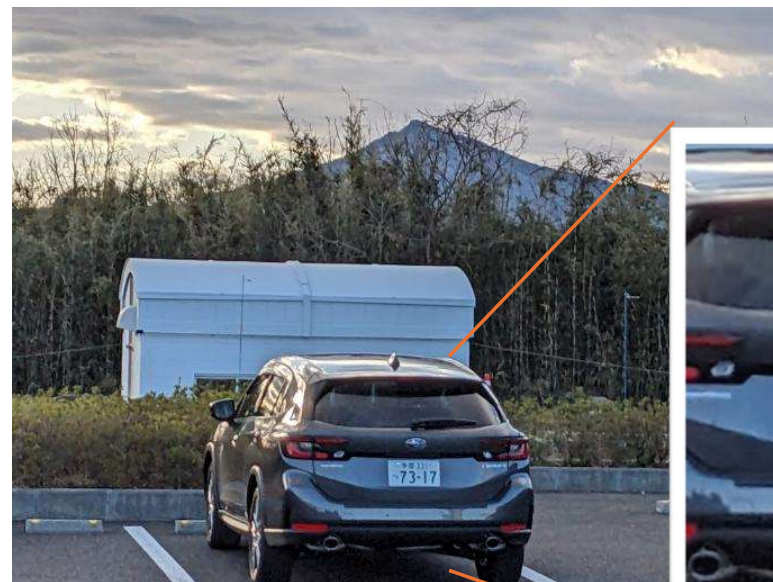


Envisaging that...

- *new countries/institutes can own and operate SLR stations.*
- *the SLR global network and the products are drastically enhanced.*
- *the SLR optical tracking/ranging technologies are more versatile.*



Omni-SLR: conveyable by a car





Omni-SLR: project status

being developed by the team of



HITOTSUBASHI UNIVERSITY

(PI: T Otsubo)



THE UNIVERSITY OF TOKYO



being supported by:

- GGOS Japan.
- the national institutes such as GSI, NIPR, NICT, Japan Coast Guard and JAXA.
- private companies in the fields of telecommunications and heavy industries.

2020-2023 First successful budget hunting (Kakenhi)
followed by various joint research projects

2020 Concept design

2021 Component tests

2022 Assembly tests

2023 Field test at NIPR Tachikawa

2024 CDP Number 7317 95 01 assigned

2024 DOMES Number 21791M008 assigned

2024 Field test at GSI Ishioka

➡ **2025** First data release to ILRS, First coordinates

2026-2027 First Antarctica SLR test at Syowa



Rooftop of
National Institute of Polar Research,
Tachikawa, Tokyo



Omni-SLR: 2025 Updates



Omni-SLR: key components (Omni-SLR A25.02 & A25.09)

CryLAS FDSS532-Q2



$\lambda = 532\text{nm}$; pulse width = 1.06 ns;
Rep. rate = 10 kHz; output = 5.7 μJ
Sync output; int/ext trigger mode

Swabian Instruments TimeTagger Ultra (Value Ed)

4 ch
(ch1 = start; ch2=stop; ch3=1PPS)
42 ps RMS
70 M events/sec



Furuno TB-1



GNSS OCXO

10 MHz: $\sim 5 \times 10^{-11}$ @ 1s
1 PPS: ~ 40 ns

Hamamatsu Photonics C11202-100



Aperture = 100 μm ; Efficiency = 70%;
Dark noise = 30 cps; No gate control

(RX) Kasai GS-200RC



Ritchey–Chrétien
 $\Phi = 203$ mm; $f = 1624$ mm

Vixen AXJ Mount (Eq.)



Converted to Alt-Az
Load: 22 kg max

Vaisala PTB-330

3 sensor barometer
0.2 hPa per sensor





Link budget calculation (simplified)

Omni-SLR
(A25.02/A25.09)



Yarragadee
(Australia)



Telescope
diameter
m ϕ

Laser
output W

0.006 mJ, 5 kHz



$$= \left[\frac{0.2}{0.76} \right]^2 \frac{0.03}{1.0} \doteq \frac{0.002}{1}$$

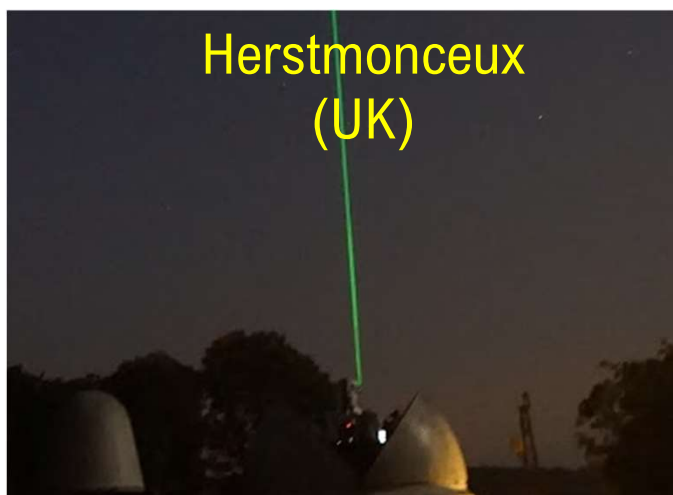


Link budget calculation (simplified)

Omni-SLR
(A25.02/A25.09)



Herstmonceux
(UK)



Telescope
diameter
m Φ

$$= \left[\frac{0.2}{0.5} \right]^2 \frac{0.03}{1.0} \doteq \frac{0.005}{1}$$

Laser
output W

0.006 mJ, 5 kHz



1 mJ, 1 kHz



**Hx applies ND 2 or 3
(1/100 or 1/1000) for
LEO, and still gets
1000 returns/NP.**

Ishioka Geodetic Observing Station of GSI

Operational
since 2016

Newly built
2024

Omni-SLR



VLBI

GNSS

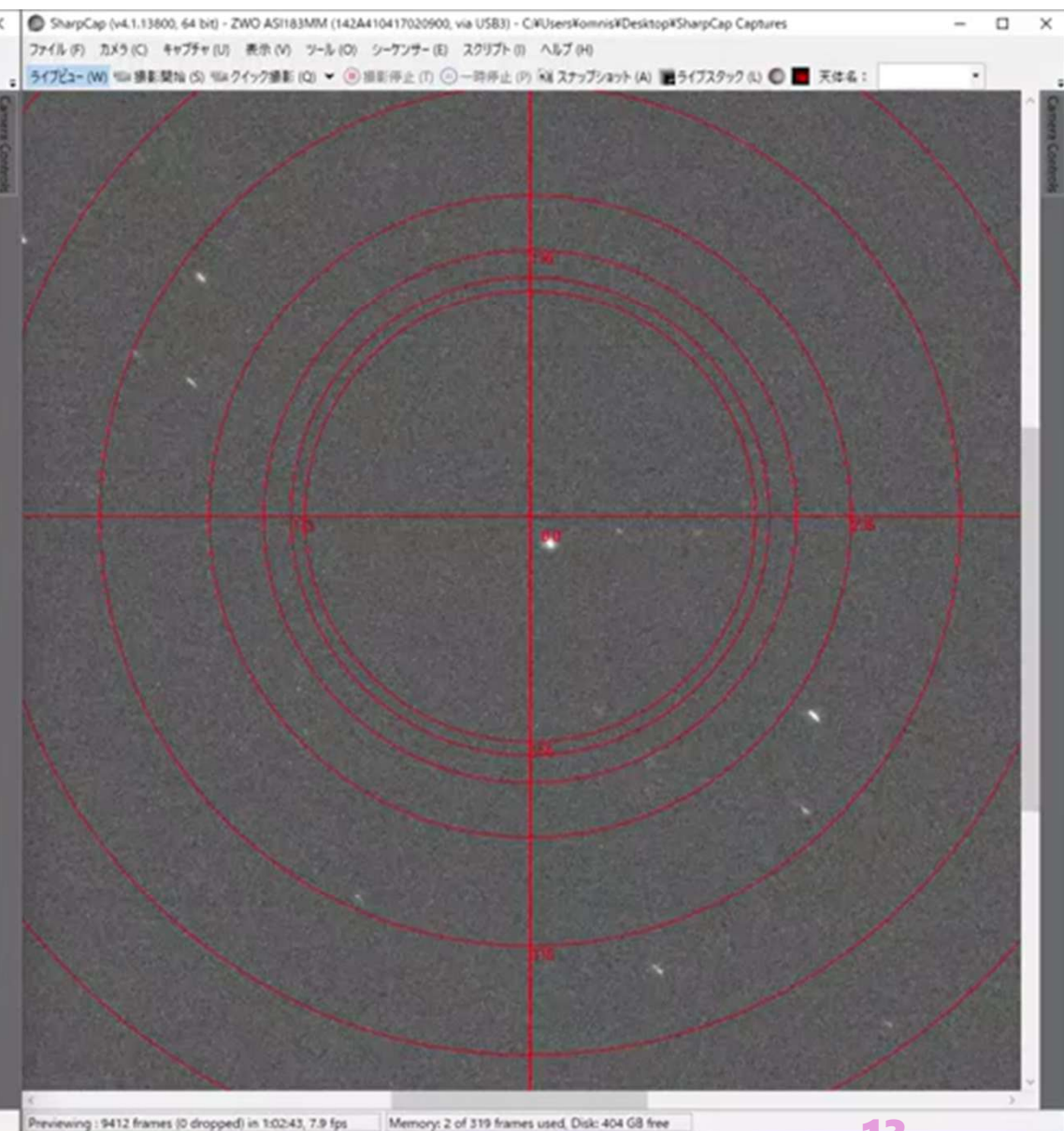
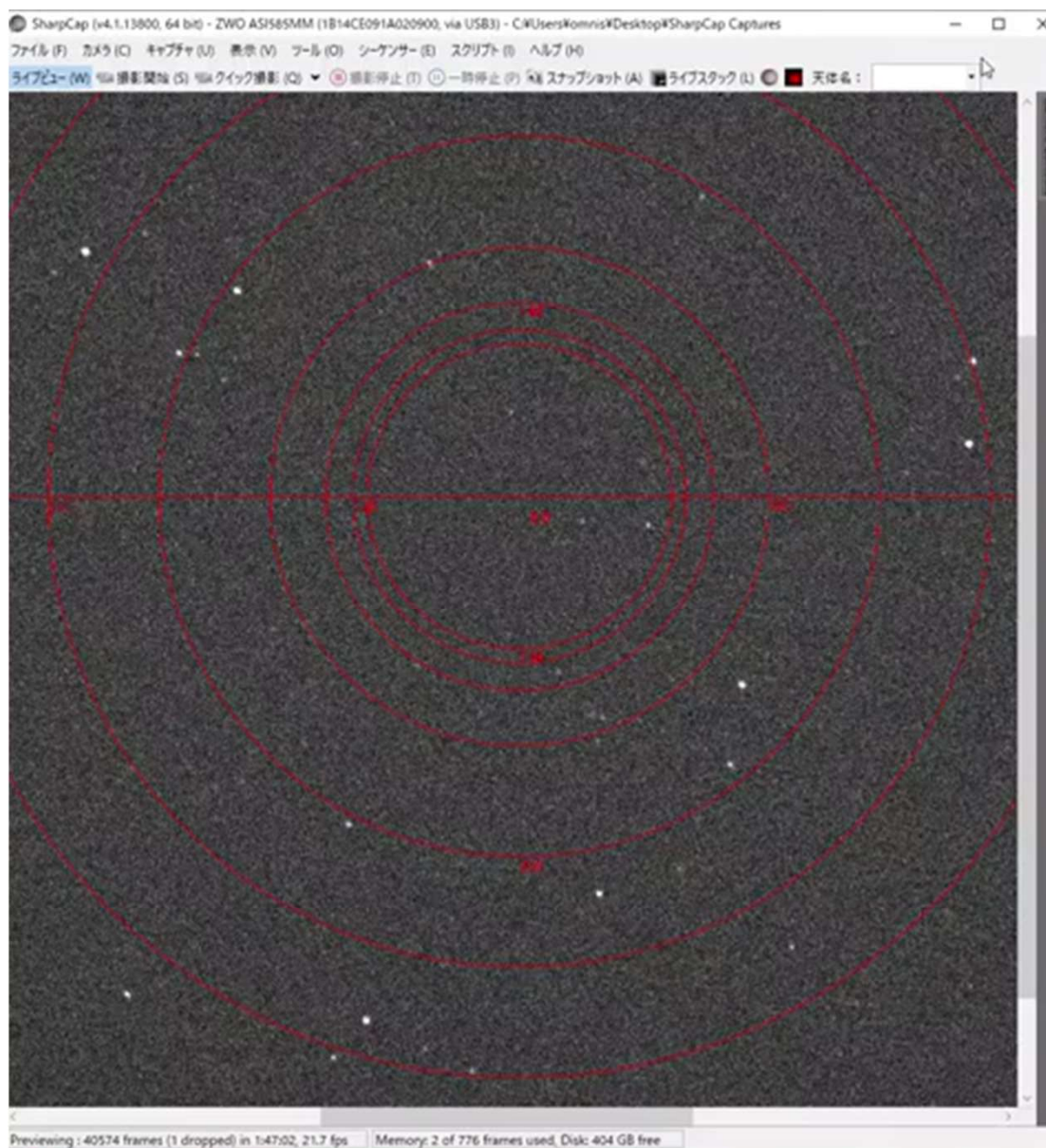
10

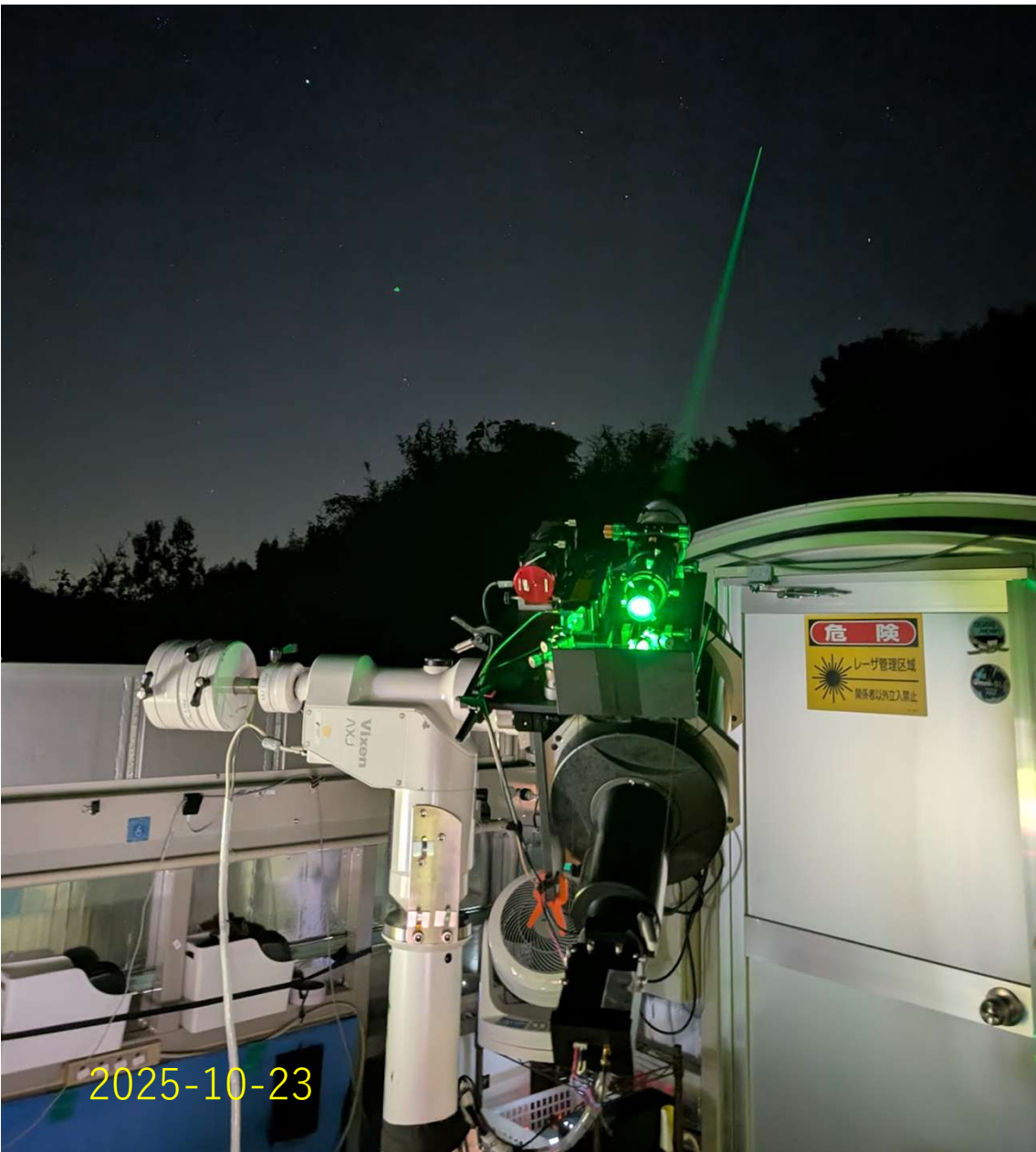


2025-10-23



2025-10-23





Component	Manufacturer	Model	Main specifications
Pulsed laser	CryLAS	FDSS532-Q2	10 kHz, 6μJ, 1.06 ns
RX Telescope	Kasai Trading	GS-200RC	203 mm diameter
Detector	Hamamatsu Photonics	C11202-100	0.1 mm aperture, SPAD
Mount	Vixen	AXJ equatorial mount	Converted to Alt-Az
Event Timer	Swabian Instruments	Time Tagger Ultra Value Edition	42 ps random jitter
GNSS Time Generator	Furuno	TB-1	
NTP Time Server	Citizen	TSV-500GP	
Barometer	Vaisala	PTB-330	3 sensors
Temperature and humidity sensor	T&D	TR-72nw	

Omni-SLR@Ishioka 13 Feb to 4 Dec, 2025

178 passes, mostly
released to EDC

Satellite	PASS	NP	FR
ajisai	41	401	177400
beaconc	27	300	78332
starlette	17	64	13200
stella	15	60	15512
swot	9	71	13740
hy2c	8	32	3692
jason3	8	24	1243
lares	8	22	1342
lageos1	6	23	2047
sentinel6a	6	27	1161
cryosat2	5	23	1423
hy2d	5	21	5236
lares2	5	7	292
saral	5	37	5541
hy2b	3	17	1415
swarmb	3	33	8401
lageos2	2	3	617

and several
satellites with
only one pass
taken.



External calibration

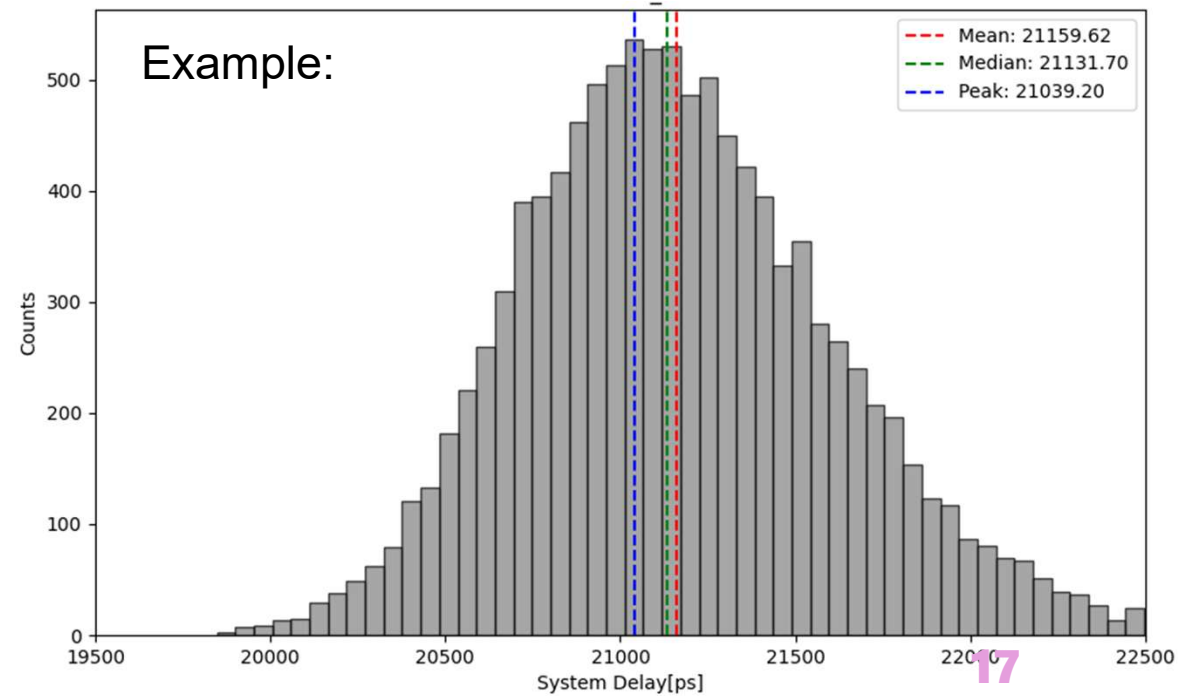


Typically 10000 returns per session

System delay : $\sim 21\text{ns}$

RMS : $\sim 435\text{ps}$

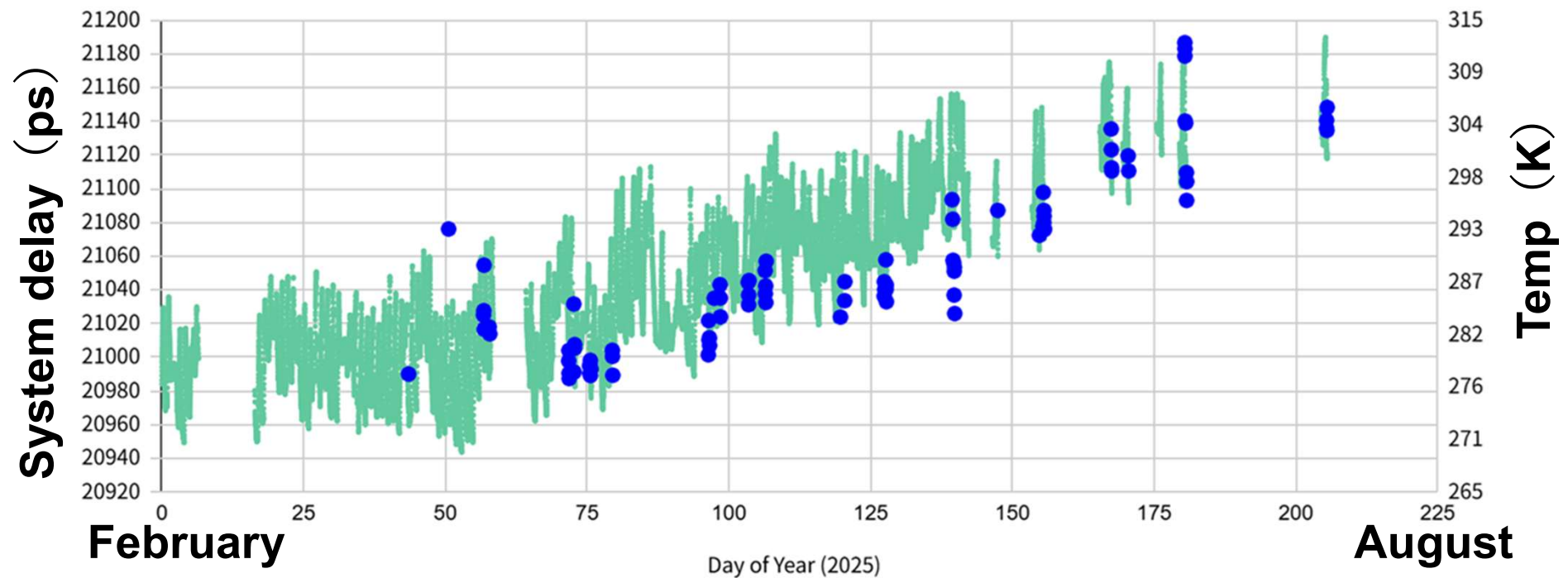
→ almost the same as the 1.06 ns FWHM laser pulse





Omni-SLR A25.02 System Delay

● Temperature ● System delay



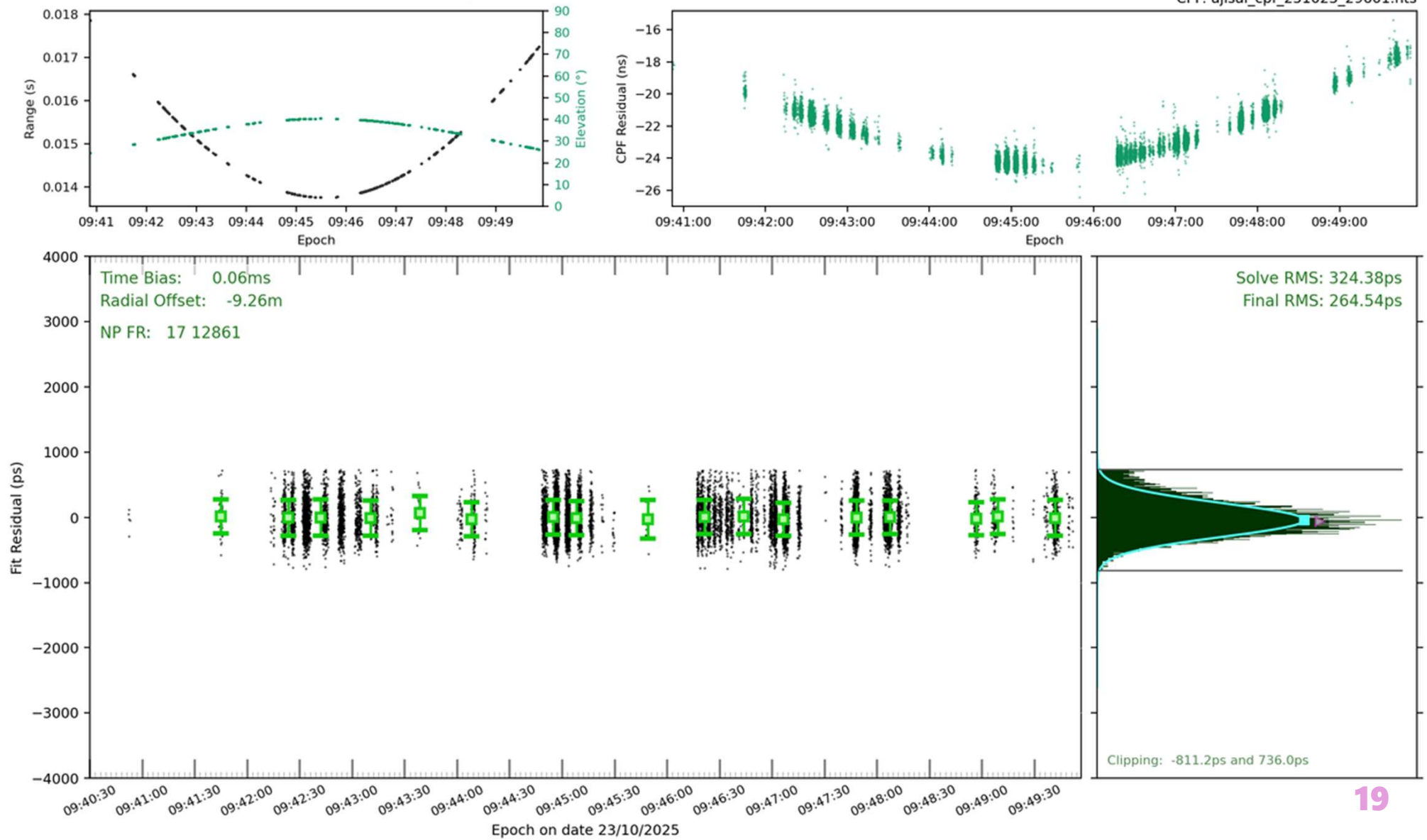
- *Within 200 ps for 6 months, ~ 5.5 ps/K: reasonably good for an open air system*
- *Pre- vs Post-cal difference: typ. < 10 ps, largest = 50 ps*
- *Short cables $\rightarrow$ Small system delay and small variation (to be stabilized more in the future)*

Satellite Laser Range data from: cpf_2ndA.dat

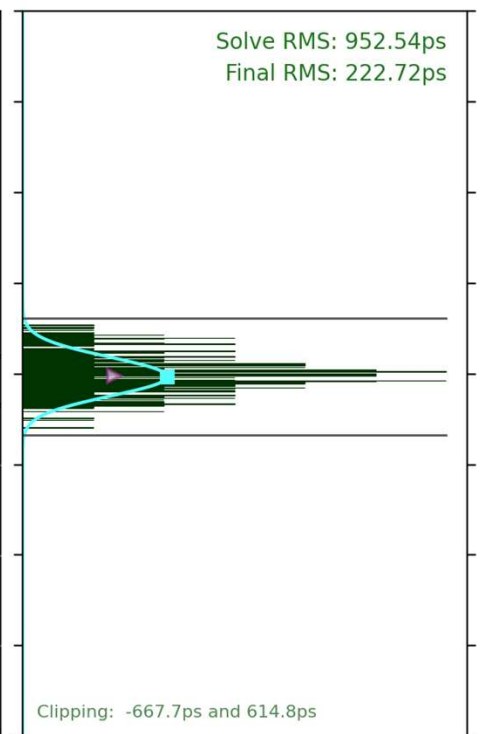
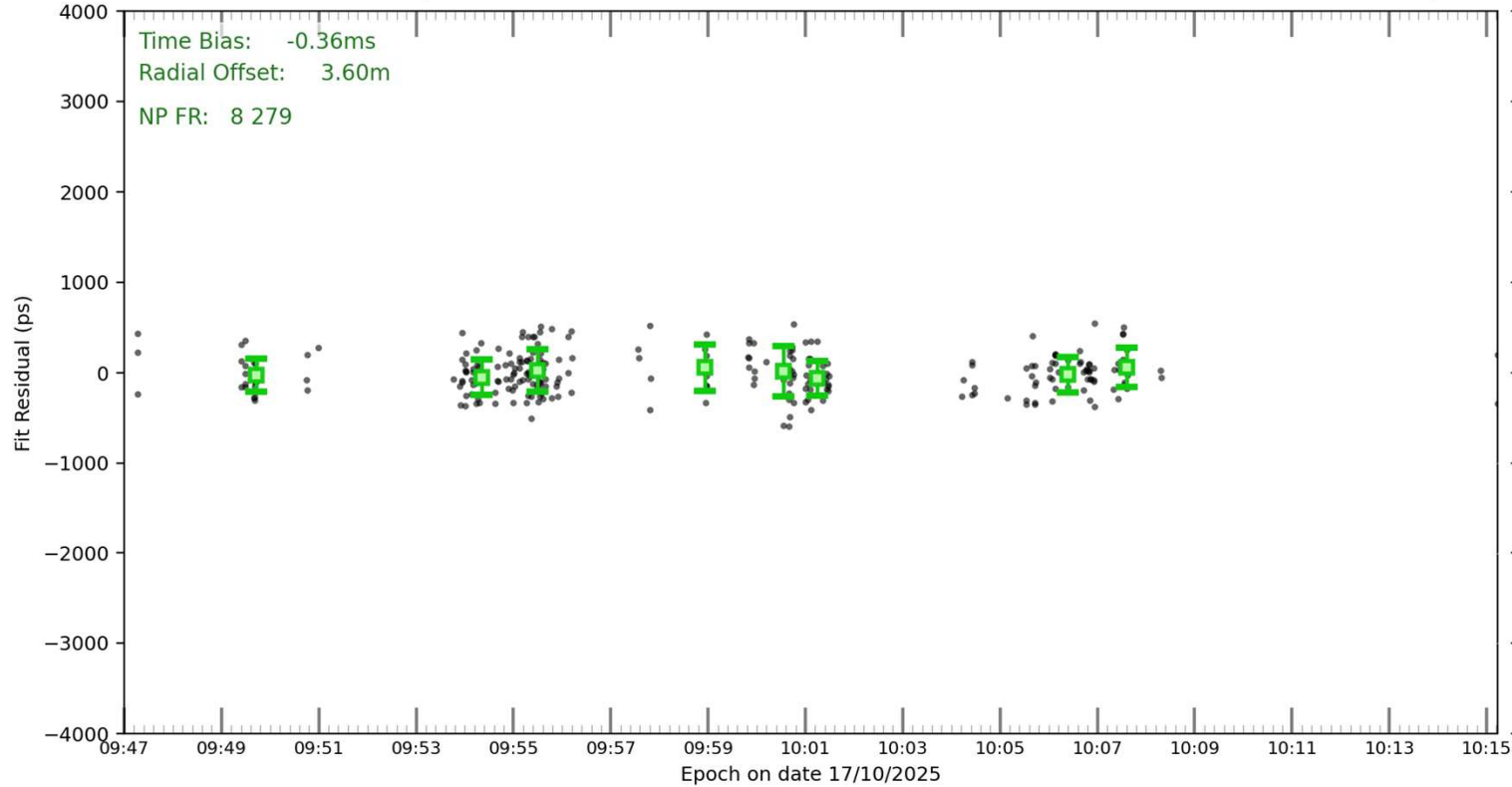
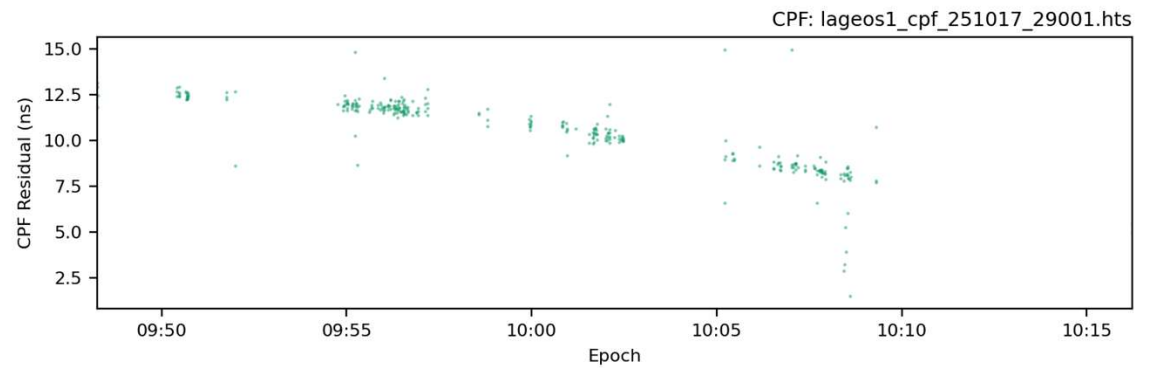
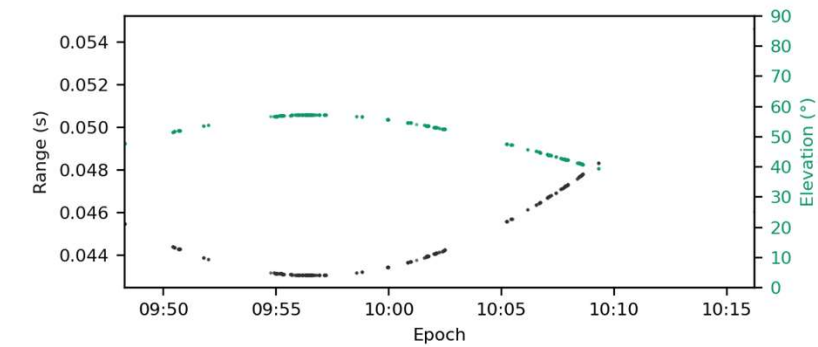
Station: 7317

Satellite: Ajisai

CPF: ajisai_cpf_251023_29601.hts



Satellite Laser Range data from: cpf_2ndA.dat
Station: 7317 Satellite: Lageos1

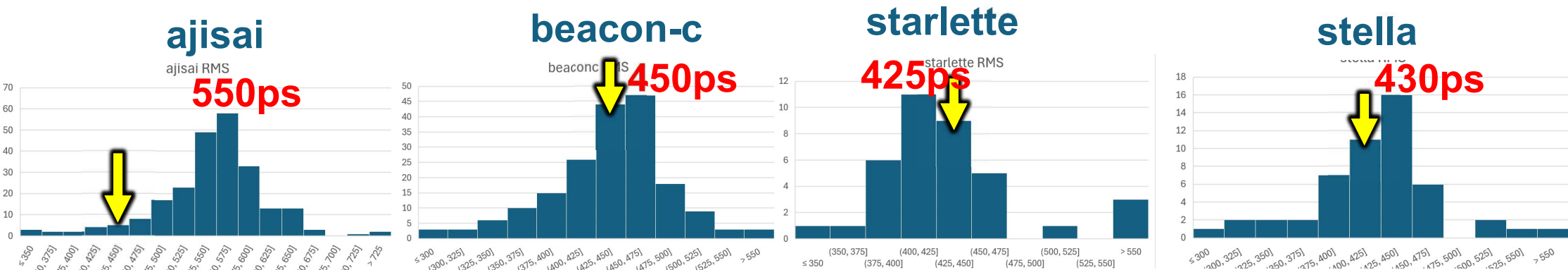




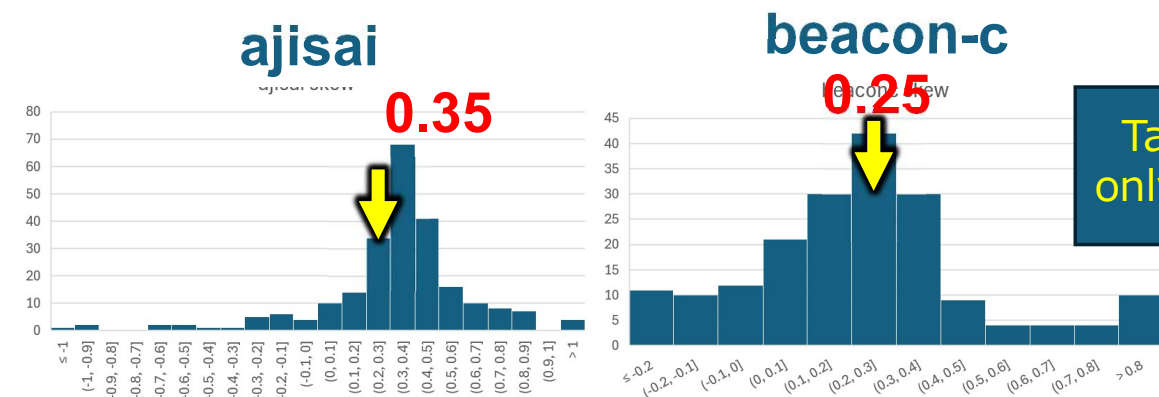
Single-shot RMS and skew

21

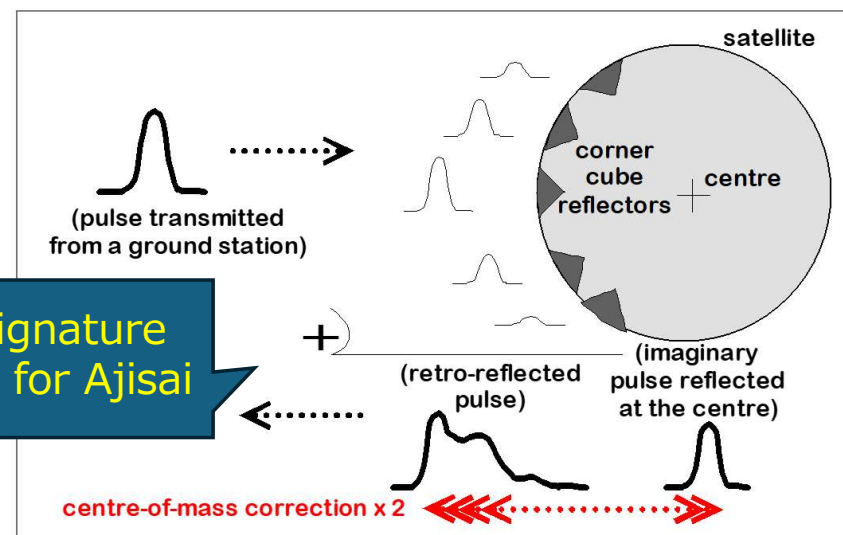
Single-shot RMS (NP basis) for LEOs Terrestrial Ranging : 435 ps



Single-shot Skew (NP basis) for LEOs
Terrestrial Ranging: 0.25



Target signature
only seen for Ajisai



Otsubo & Appleby 2003 JGR

SLR Data → First solution



First estimate of station coordinates

Using our own c5++ software

3 satellites (Ajisai, Starlette, Stella) combined

Applying Ishioka's velocity and PSD of ITRF2020-u2024:

7317 ISHL

$X = 3959591.0713 \pm 0.0133 \text{ m}$

$Y = 3296886.2143 \pm 0.0098 \text{ m}$

$Z = 3747021.1178 \pm 0.0130 \text{ m}$

$\text{Range bias} = 0.0005 \pm 0.0128 \text{ m}$

→ To Do: Include LAGEOS-1, -2, LARES, LARES-2

→ To Do: Compare them with GSI's local tie (2025.10)

QC'ed by ourselves



Multi-Satellite Bias Analysis Report v2 for Worldwide Satellite Laser Ranging Stations being updated every 6 hours!

Latest Analysis Report: >> [from 12h UTC, 26 Oct 2025 to 12h UTC, 09 Nov 2025](#) (updated 14:15 UTC, 9 Nov 2025)

Stations with high productivity

sat	orbit fit WRMS # pass/# NP in mm	1st site(ID) # pass/# NP	2nd site(ID) # pass/# NP	3rd site(ID) # pass/# NP
Lageos-1	9 309 / 2599	Yarragadee (7090) 35/332	Potsdam (7841) 32/171	Graz (7839) 24/301
Lageos-2	10 204 / 1824	Yarragadee (7090) 41/478	Zimmerwald (7810) 19/250	Potsdam (7841) 18/104
Etalon-1	12 44 / 257	Wettzell (8834) 10/87	Yarragadee (7090) 10/58	Graz (7839) 6/18
Etalon-2	9 46 / 259	Wettzell (8834) 8/72	Graz (7839) 7/47	Wettzell (7827) 6/40
Ajisai	30 337 / 4526	Yarragadee (7090) 62/693	Wettzell (8834) 26/597	Zimmerwald (7810) 25/404
Lares	14 279 / 2986	Yarragadee (7090) 38/398	Potsdam (7841) 38/234	Graz (7839) 28/374
Lares-2	8 226 / 2077	Yarragadee (7090) 27/268	Potsdam (7841) 23/138	Zimmerwald (7810) 21/381
Starlette	25 334 / 2890	Yarragadee (7090) 59/563	Graz (7839) 29/367	Changchun (7237) 27/160
Stella	25 219 / 1735	Yarragadee (7090) 35/330	Graz (7839) 20/190	Potsdam (7841) 17/90

and more satellites (GNSS and LEO) are included in the reports!!

Archive: (each covers 14 days from the date) [2024](#) [2023](#) [2022](#) [2021](#) [2020](#) [2019](#) [2018](#) [2017](#) [2016](#) [2015](#) [2014](#) [2013](#) [2012](#) v1: Year [2011](#) [20](#)

Oct 2025	Sep 2025	Aug 2025	Jul 2025	Jun 2025	May 2025	Apr 2025	Mar 2025	Feb 2025	Jan 2025
26 (00 06 12 18)	30 (00 06 12 18)	31 (00 06 12 18)	31 (00 06 12 18)	30 (00 06 12 18)	31 (00 06 12 18)	30 (00 06 12 18)	31 (00 06 12 18)	28 (00 06 12 18)	31 (00 06 12 18)
25 (00 06 12 18)	29 (00 06 12 18)	30 (00 06 12 18)	30 (00 06 12 18)	29 (00 06 12 18)	30 (00 06 12 18)	29 (00 06 12 18)	30 (00 06 12 18)	27 (00 06 12 18)	30 (00 06 12 18)
24 (00 06 12 18)	28 (00 06 12 18)	29 (00 06 12 18)	29 (00 06 12 18)	28 (00 06 12 18)	29 (00 06 12 18)	28 (00 06 12 18)	29 (00 06 12 18)	26 (00 06 12 18)	29 (00 06 12 18)
23 (00 06 12 18)	27 (00 06 12 18)	28 (00 06 12 18)	28 (00 06 12 18)	27 (00 06 12 18)	28 (00 06 12 18)	27 (00 06 12 18)	28 (00 06 12 18)	25 (00 06 12 18)	28 (00 06 12 18)
22 (00 06 12 18)	26 (00 06 12 18)	27 (00 06 12 18)	27 (00 06 12 18)	26 (00 06 12 18)	27 (00 06 12 18)	26 (00 06 12 18)	27 (00 06 12 18)	24 (00 06 12 18)	27 (00 06 12 18)

QC'ed by ourselves

(only spherical satellites)

#

7317 = ISHIOKA

#

sat site date time dur rb mm error tb us error prec bad total rms pres temp hum sdelay

LAG1 7317 2025/10/17 09:50 17 -58 (25) 8.3 (22.7) 12 0 / 8 65 1012.1 289.5 84 3088

AJI1 7317 2025/10/17 11:03 6 5 (9) 8.5 (3.9) 3 0 / 13 80 1012.1 289.2 84 3088

LARS 7317 2025/10/17 12:36 0 116 (68) -----.- (----.-) 23 0 / 3 55 1012.1 289.2 81 3089

STRL 7317 2025/10/17 17:03 0 -186 (13) -----.- (----.-) 2 0 / 3 58 1009.5 287.8 91 3086

LAR2 7317 2025/10/17 17:59 0 ----- (----) -----.- (----.-) ----- 0 / 1 60 1009.0 287.2 91 3087

AJI1 7317 2025/10/23 09:41 7 22 (7) 10.3 (3.1) 7 0 / 17 80 1013.6 285.5 74 3093

STEL 7317 2025/10/23 11:34 0 25 (17) -----.- (----.-) 2 0 / 2 64 1014.0 285.1 74 3092

STRL 7317 2025/10/23 13:31 0 ----- (----) -----.- (----.-) ----- 0 / 1 41 1013.8 283.9 78 3091

LAG1 7317 2025/10/23 15:37 2 -72 (41) -----.- (----.-) 10 0 / 2 59 1013.0 283.5 76 3092

LAR2 7317 2025/11/13 16:17 1 -30 (22) -----.- (----.-) 3 0 / 2 75 998.1 282.7 70 3088

LAG2 7317 2025/11/13 16:53 1 -5 (27) -----.- (----.-) 4 0 / 2 65 998.4 282.5 73 3088

LARS 7317 2025/11/13 17:33 0 ----- (----) -----.- (----.-) ----- 0 / 1 62 998.1 281.9 74 3087

AJI1 7317 2025/11/13 17:49 7 93 (26) -14.1 (11.1) 12 0 / 11 83 998.4 281.7 74 3086

LAG1 7317 2025/11/13 18:24 27 -16 (11) -9.9 (7.7) 10 0 / 8 60 998.3 282.1 69 3086

LARS 7317 2025/11/13 19:26 1 13 (32) -----.- (----.-) 25 0 / 3 73 998.5 281.9 70 3087

AJI1 7317 2025/11/13 19:51 9 -1 (28) -19.3 (7.8) 15 0 / 14 87 998.4 281.8 70 3087

- Single-shot RMS: ~ 80 mm for Ajisai and ~ 65 mm for others
- NP RMS (fitted per pass): < 10 mm for densely-tracked satellites
- Range bias for LEO spherical satellites: comparable to the existing stations. 24

QC'ed by ourselves

#

7317 = ISHIOKA

#	sat	site	date	time	dur	rb mm	error	tb us	error	prec	bad	total	rms	pres	temp	hum	sdelay
SARL	7317		2025/10/17	09:31	2	-50 (	16)	-----.-	(----.-)	5	0 /	10	58	1011.8	290.4	80	3090
LAG1	7317		2025/10/17	09:50	17	-58 (	25)	8.3	(22.7)	12	0 /	8	65	1012.1	289.5	84	3088
AJI1	7317		2025/10/17	11:03	6	5 (	9)	8.5	(3.9)	3	0 /	13	80	1012.1	289.2	84	3088
LARS	7317		2025/10/17	12:36	0	116 (	68)	-----.-	(----.-)	23	0 /	3	55	1012.1	289.2	81	3089
CRY2	7317		2025/10/17	15:31	0	1405 (	66)	-----.-	(----.-)	9	0 /	2	53	1010.4	289.2	83	3088
BECC	7317		2025/10/17	15:55	0	-----	(-----)	-----.-	(----.-)	-----	0 /	1	56	1010.3	287.4	92	3086
STRL	7317		2025/10/17	17:03	0	-186 (	13)	-----.-	(----.-)	2	0 /	3	58	1009.5	287.8	91	3086
BECC	7317		2025/10/17	17:50	1	352 (	54)	-----.-	(----.-)	13	0 /	6	68	1009.1	287.6	88	3087
LAR2	7317		2025/10/17	17:59	0	-----	(-----)	-----.-	(----.-)	-----	0 /	1	60	1009.0	287.2	91	3087
AJI1	7317		2025/10/23	09:41	7	22 (	7)	10.3	(3.1)	7	0 /	17	80	1013.6	285.5	74	3093
BECC	7317		2025/10/23	11:23	3	52 (	29)	-8.0	(17.3)	8	0 /	9	65	1013.8	285.4	73	3092
STEL	7317		2025/10/23	11:34	0	25 (	17)	-----.-	(----.-)	2	0 /	2	64	1014.0	285.1	74	3092
BECC	7317		2025/10/23	13:22	0	73 (	74)	-----.-	(----.-)	10	0 /	2	61	1013.9	284.1	77	3091
STRL	7317		2025/10/23	13:31	0	-----	(-----)	-----.-	(----.-)	-----	0 /	1	41	1013.8	283.9	78	3091
BECC	7317		2025/10/23	15:09	6	48 (	35)	24.9	(10.2)	10	0 /	9	61	1013.0	284.6	74	3092
CRY2	7317		2025/10/23	15:26	3	-50 (	705)	-----.-	(----.-)	113	0 /	2	43	1012.9	283.9	75	3092
LAG1	7317		2025/10/23	15:37	2	-72 (	41)	-----.-	(----.-)	10	0 /	2	59	1013.0	283.5	76	3092

➡ **Single-shot RMS: ~ 80 mm for Ajisai and ~ 65 mm for others**

➡ **NP RMS (fitted per pass): < 10 mm for densely-tracked satellites**

➡ **Range bias for LEO spherical satellites: comparable to the existing stations.**

Omni-SLR: Ongoing/Future plans



Omni-SLR: possible upgrades (Omni-SLR **A-Next**)

Teem BNG-15E-000

l-a: Controller



l-b: Laser head



$\lambda = 532\text{nm}$; pulse width = 0.75 ns;
Rep. rate = 22 kHz; output = 17 μJ
Sync output; int/ext trigger mode

Swabian Instruments TimeTagger Ultra (Value Ed)

4 ch
(ch1 = start; ch2=stop; ch3=1PPS)
42 ps RMS
70 M events/sec



FreqTime FT-RB or?



GNSS Rb
 $2 \times 10^{-11} @ 1\text{s}$
 $2 \times 10^{-12} @ 100\text{ s}$

Hamamatsu Photonics C11202-100



Aperture = 100 μm ; Efficiency = 70%;
Dark noise = 30 cps; No gate control

(RX) Kasai GS-200RC



Ritchey–Chrétien
 $\Phi = 203\text{ mm}$; $f = 1624\text{ mm}$

Hobym Crux200MF



Alt-Az mode
Load: 30 kg max

Vaisala PTB-330

3 sensor barometer
0.2 hPa per sensor



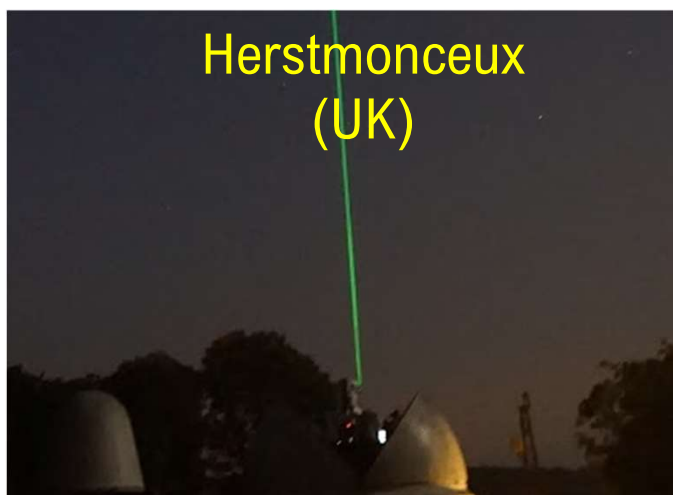


Link budget calculation (simplified)

Omni-SLR
(A25.02/A25.09)



Herstmonceux
(UK)



Telescope
diameter
m Φ

$$= \left[\frac{0.2}{0.5} \right]^2 \frac{0.03}{1.0} \doteq \frac{0.005}{1}$$

Laser
output W

0.006 mJ, 5 kHz



1 mJ, 1 kHz

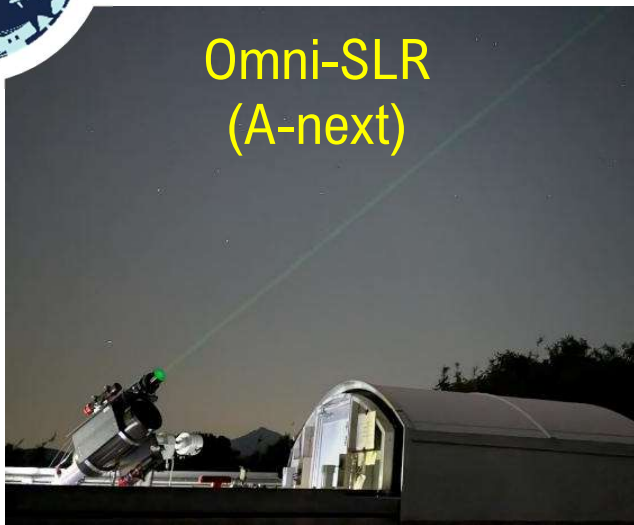


Hx applies ND 2 or 3
(1/100 or 1/1000) for
LEO, and still gets
1000 returns/28.

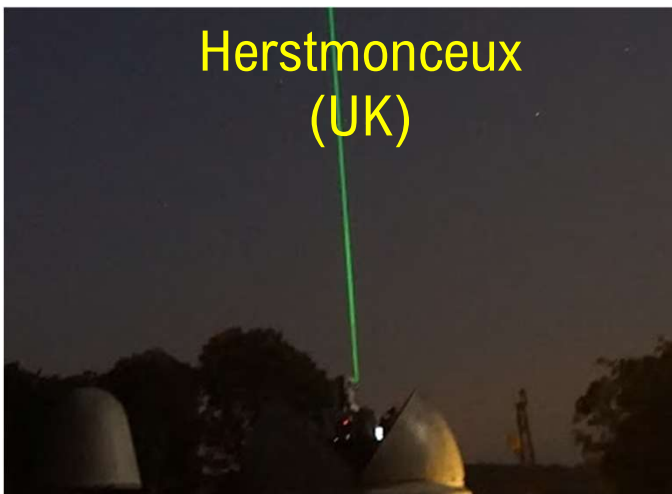


Link budget calculation (simplified)

Omni-SLR
(A-next)



Herstmonceux
(UK)



Telescope diameter
m Φ

Laser
output W

0.15 mJ, 20 kHz

(just arrived)

0.15



~~0.03~~

1.0

0.025

1

$$= \left[\frac{0.2}{0.5} \right]^2$$

$\div$



1 mJ, 1 kHz



Additional
budget
options



Antarctic test approved

NIPR Exploratory Research Project

“Development of a compact SLR system for the first antarctica experiment”

3 years: FY2024 to FY2026. Syowa: 69S 35E.

FY2024-FY2025

Preliminary tests in Japan.

FY2025

A roll-off roof container to be installed at Syowa (if possible).

FY2026 (JARE68)

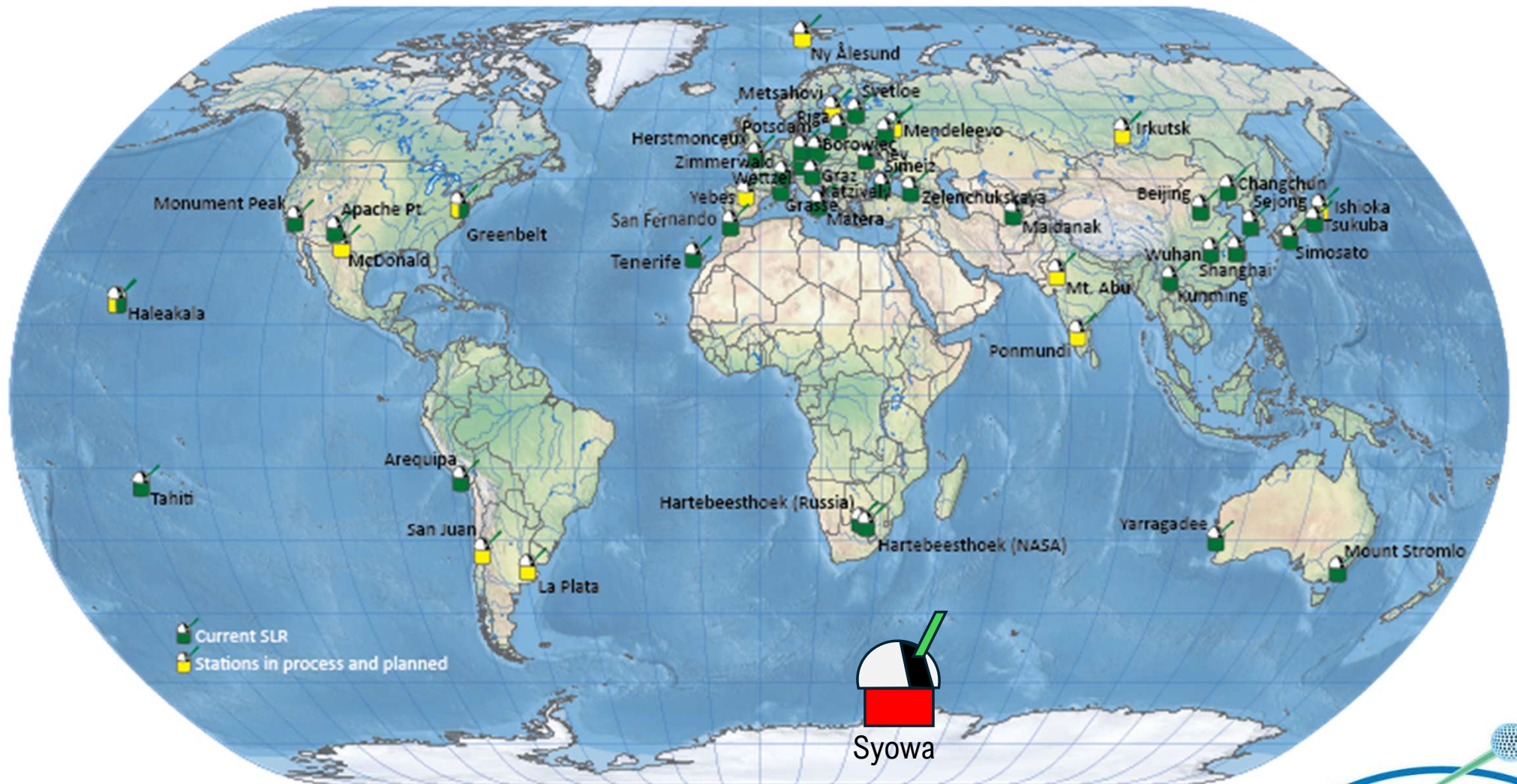
Shipped to Syowa (Nov 2026). **SLR test at Syowa (Jan-Feb 2027).**



→
Syowa's container
= 3.0 m x 2.4 m



Current and Planned ILRS Network



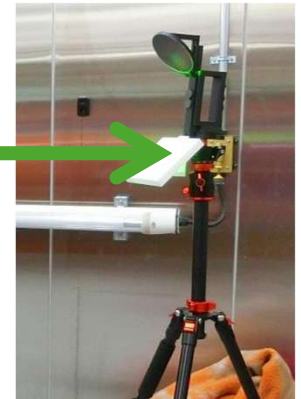
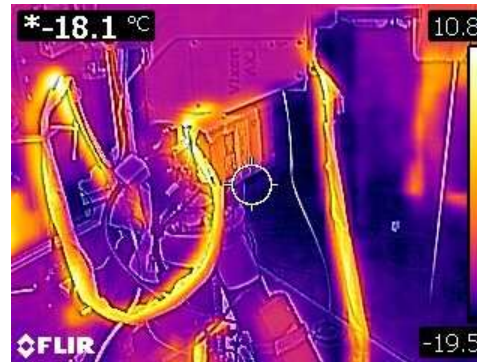
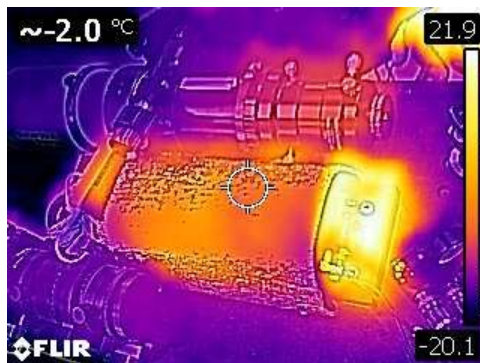
Exploratory Research Projects, AH1004 (FY2024-2026)

✓ Roll-off roof



Being shipped to
Syowa JARE67.

✓ Low-temperature test

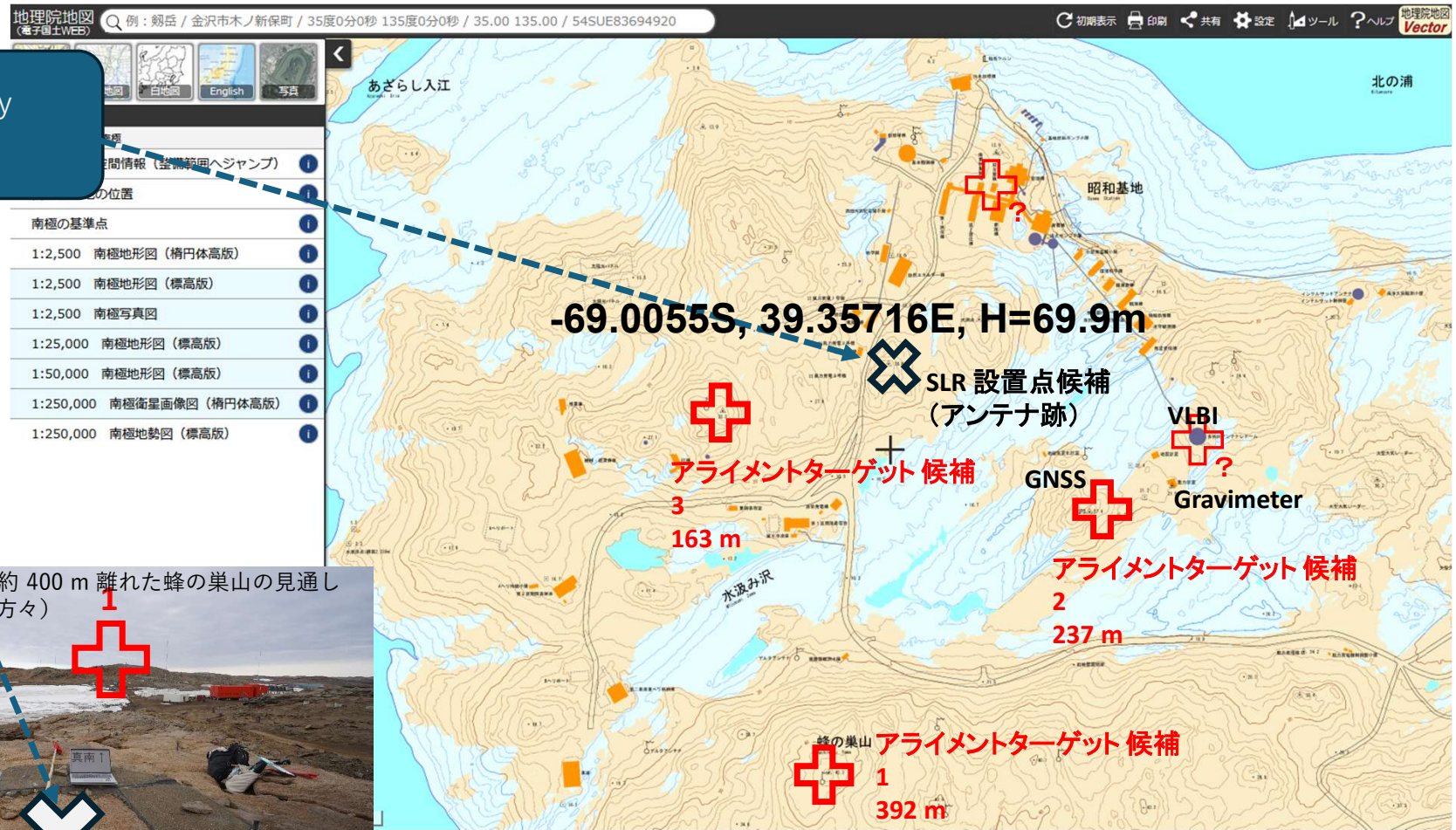


□ Daytime-only operation (being tested@Ishioka)

Antarctica's First SLR attempts, Jan to Feb 2027 (JARE68)

(After JARE68, the Omni-SLR equipment will be removed and stored, awaiting the next project.)

Site survey done by
JARE59 and 66





Conclusions & Ongoing+Future projects

A very low-cost SLR works! 150+ LEO passes and several LAG passes.

- **Initial coordinates of Ishioka estimated.**
- **Random errors tolerated; systematic errors suppressed.**
- **Limits: optical link, tracking stability,...**

✓ **2025.10 Local tie incl Omni-SLR (by GSI)**

✓ **2025.11 1st paper submitted to J Geodesy**

☐ **2025 4Q Installation of a new laser (Teem)**

☐ **2026 1Q cm coordinates solution to be compared with GSI's local tie**

☐ **2026 1Q Daytime observation (SLR and StarCal)**

☐ **2026 1Q-2Q Solid LAGEOS tracking → Sub-cm coordinates.**

☐ **2026 3Q-4Q First SLR in Antarctica ("S set") ← ILRS site registration soon.**

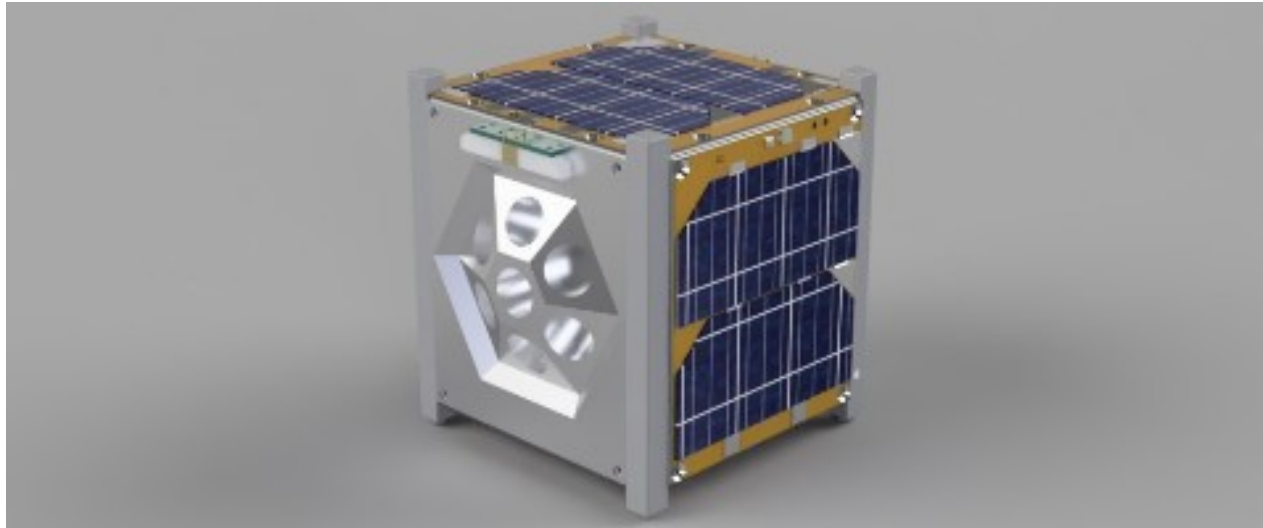
☐ **2027+ Remote operation, Automation,...**



Happy Holidays
and
Happy New Year!



Difficulties in observing the e-kagaku satellite using SLR JAXA Shinichi Nakamura



First and foremost, we would like to express our gratitude for your cooperation with the SLR observations of the ekagaku1 satellite. We also apologize for the lack of response from the ekagaku1 satellite. Today, we will explain the cause of this issue and share some points to keep in mind during SLR observations.

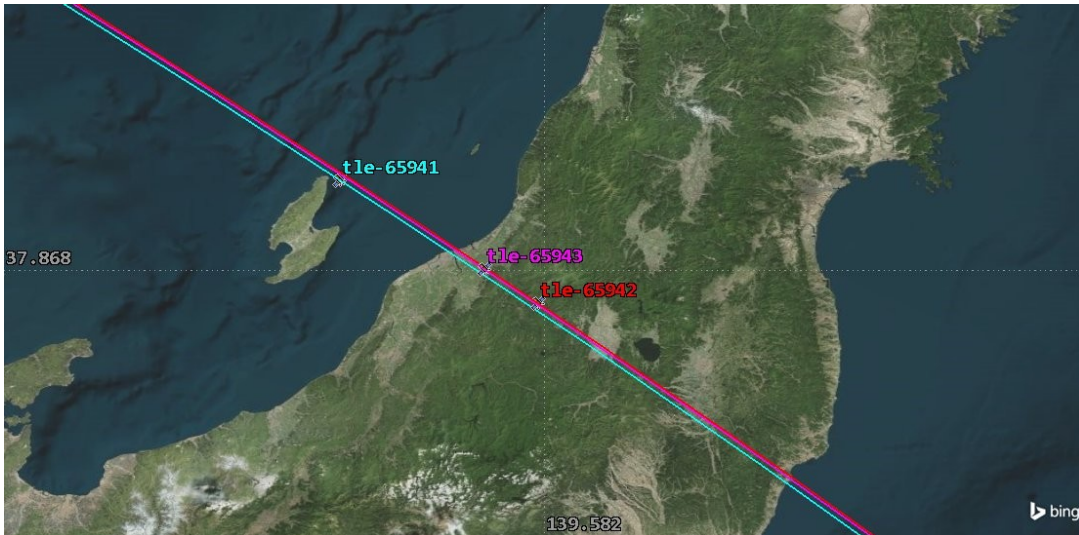
Cause 1: TLE Misassignment

On October 10th, three small satellites were deployed from the ISS.

Unfortunately, radio communication was not established, so we made a CPF for the ekagaku1 satellite using TLE from November 4th to 8th.

According TLE, 65941 Clover, 65942 ekagaku1, and 65943 BOTAN.

However, on November 11th, with the cooperation of the amateur radio community, it was determined that TLE-65942 was the BOTAN satellite.



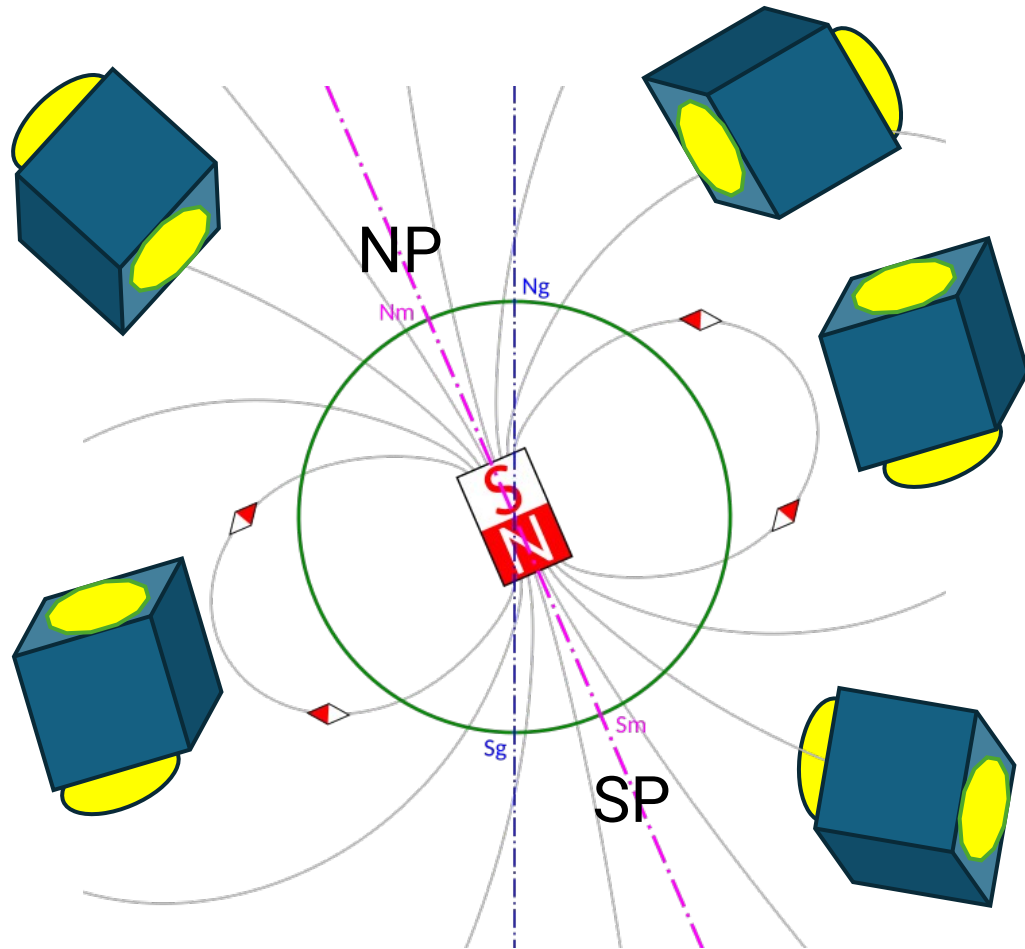
On Oct 15th

The direction of travel is from top left to bottom right. The flight order, from the front, is 65942, 65943, 95941.

Also, since the Clover satellite is a 2U satellite and the ekagaku1 satellite is a 1U satellite, we determined that TLE-65943 was the ekagaku1 satellite based on a radar cross section evaluation using the JAXA SSA radar. So, we made the CPF by considering TLE-65943 as the ekagaku1 satellite. (The CPF was based on the orbit determined by the JAXA SSA radar, not the TLE.)

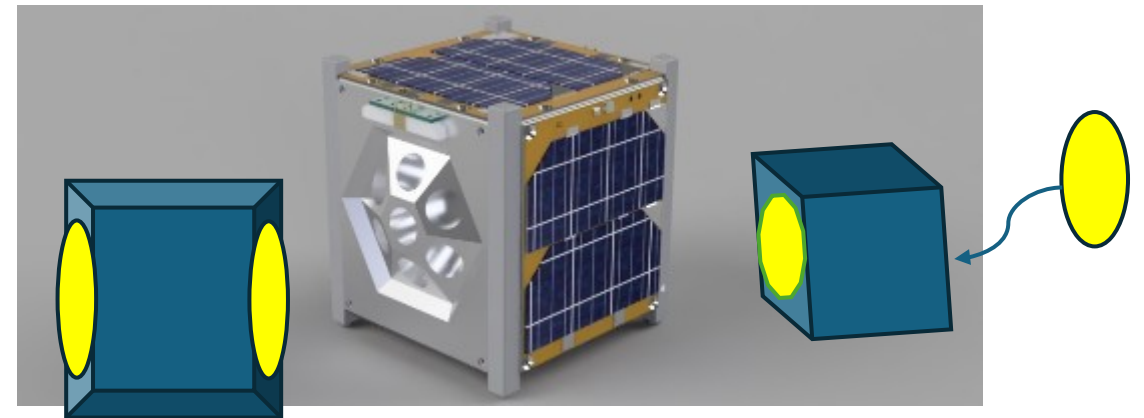
However, we have not yet obtained return signal to the SLR laser.

Cause 2: Satellite Design



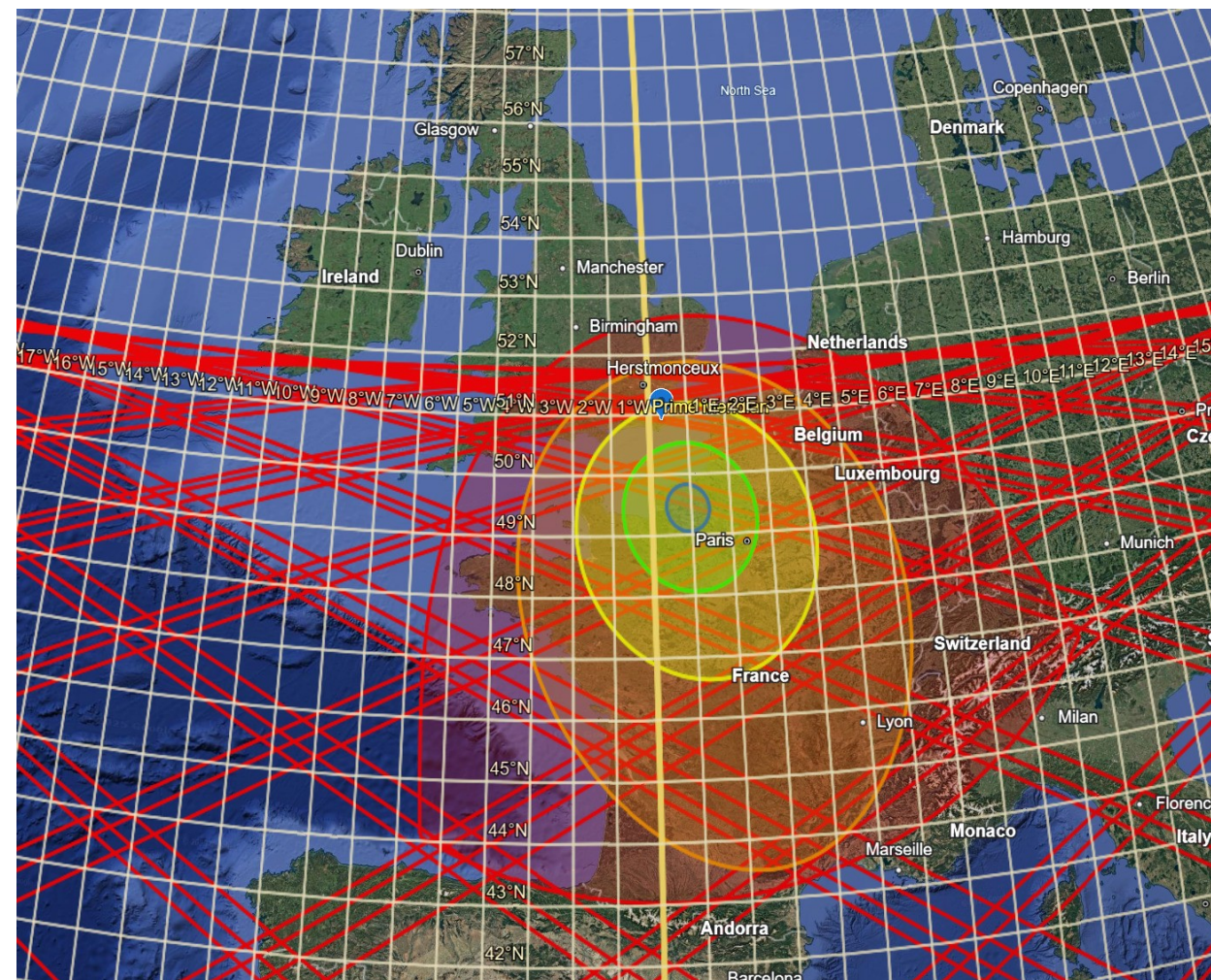
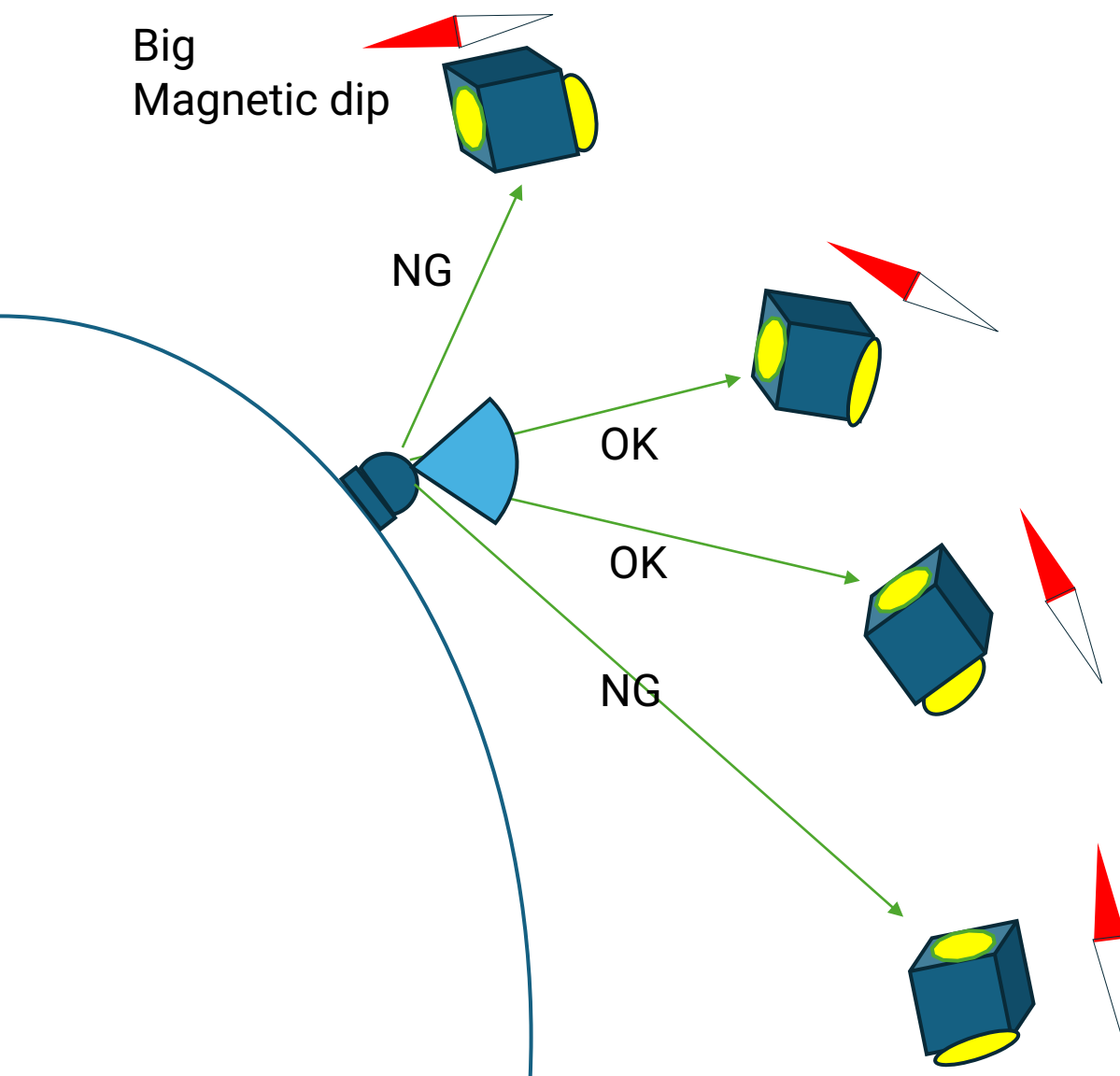
Last week, I discussed the ekagaku project and the cause of the SLR failure. There was insufficient communication between JAXA and the junior and senior high school students who developed the ekagaku1 satellite.

The reflector attached to the ekagaku1 satellite was not facing toward the Earth. The reflector faces north, along the Earth's magnetic field lines.



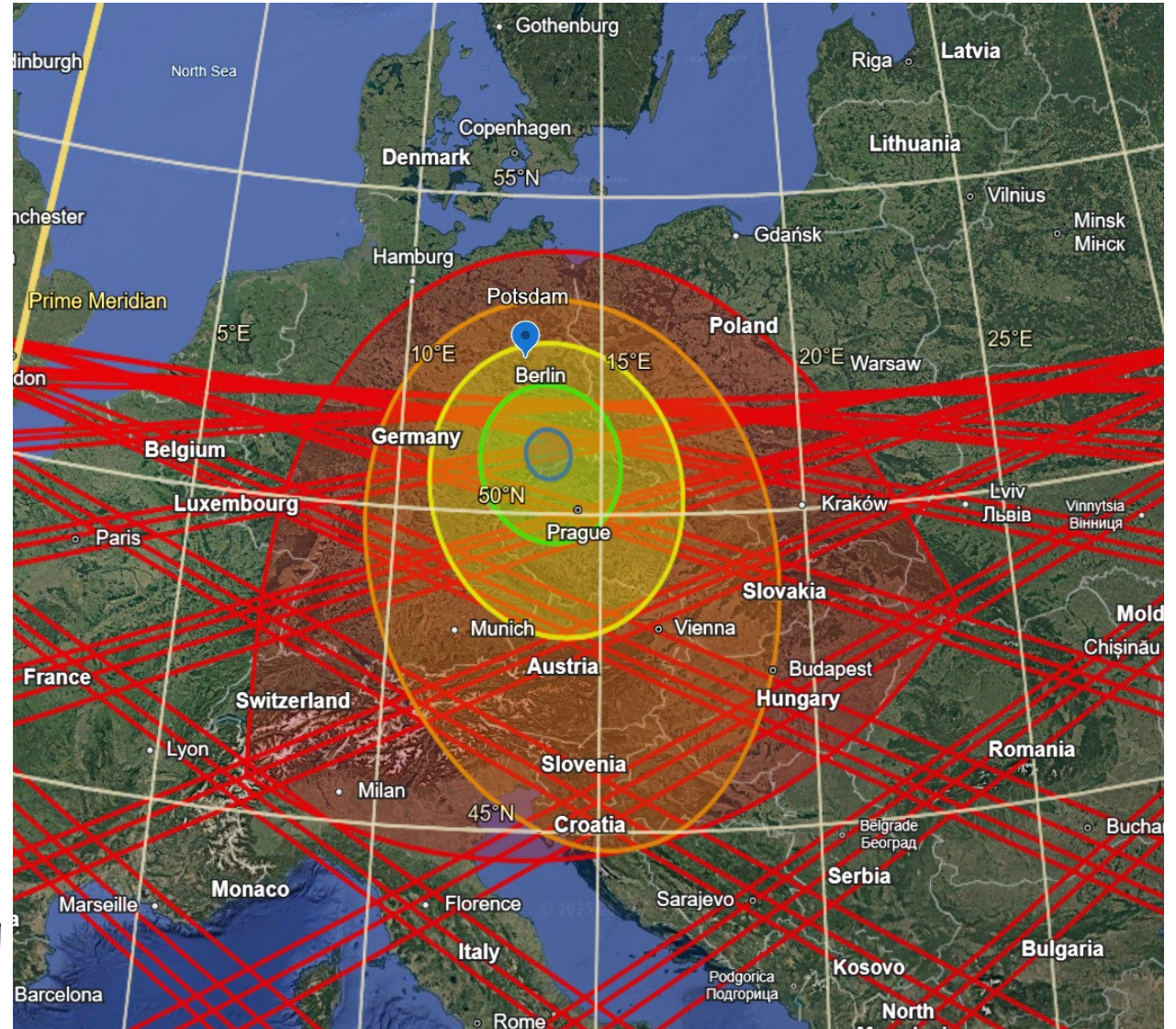
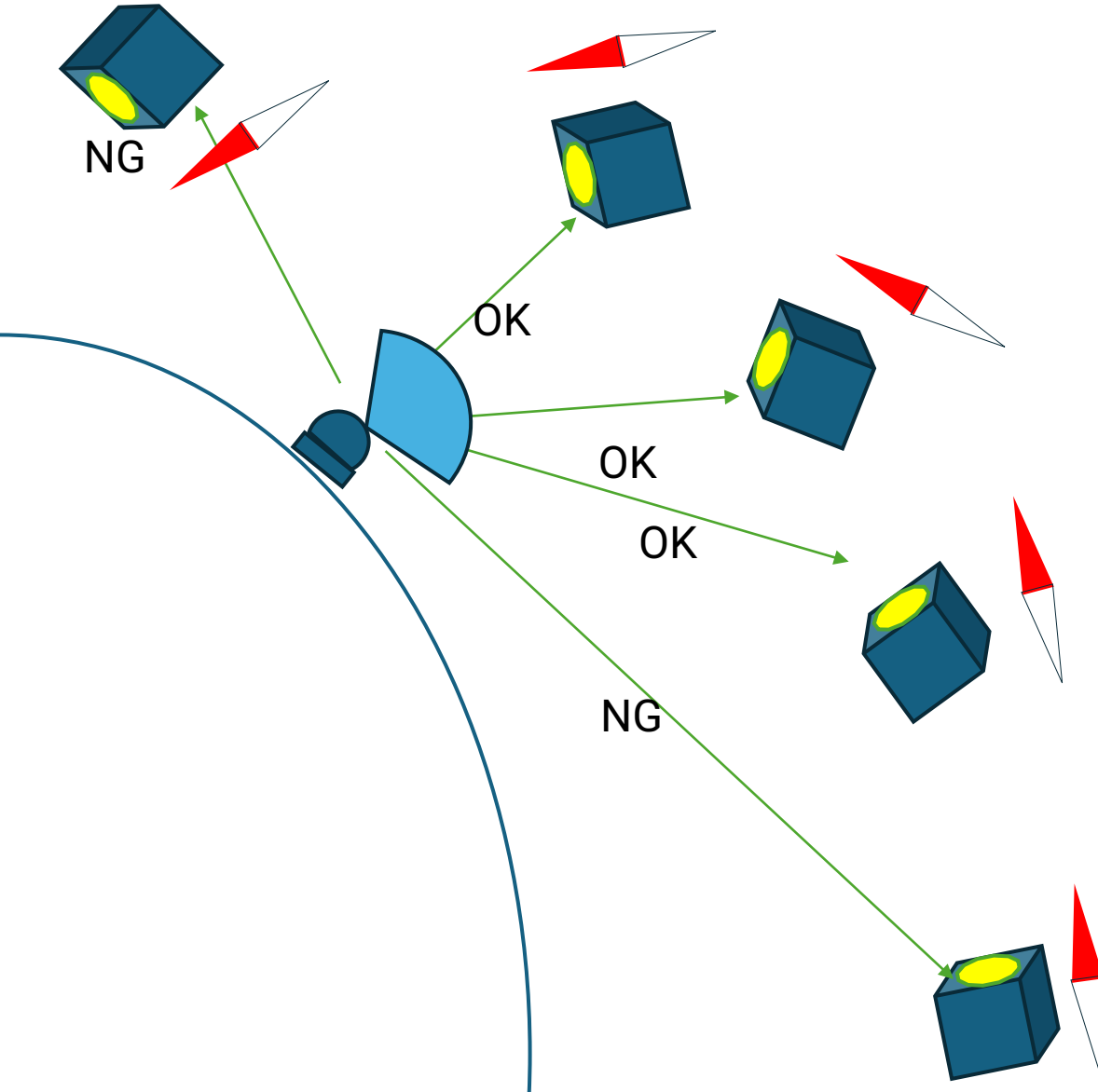
High-latitude stations (Latitude < 51)

northern hemisphere SLR station



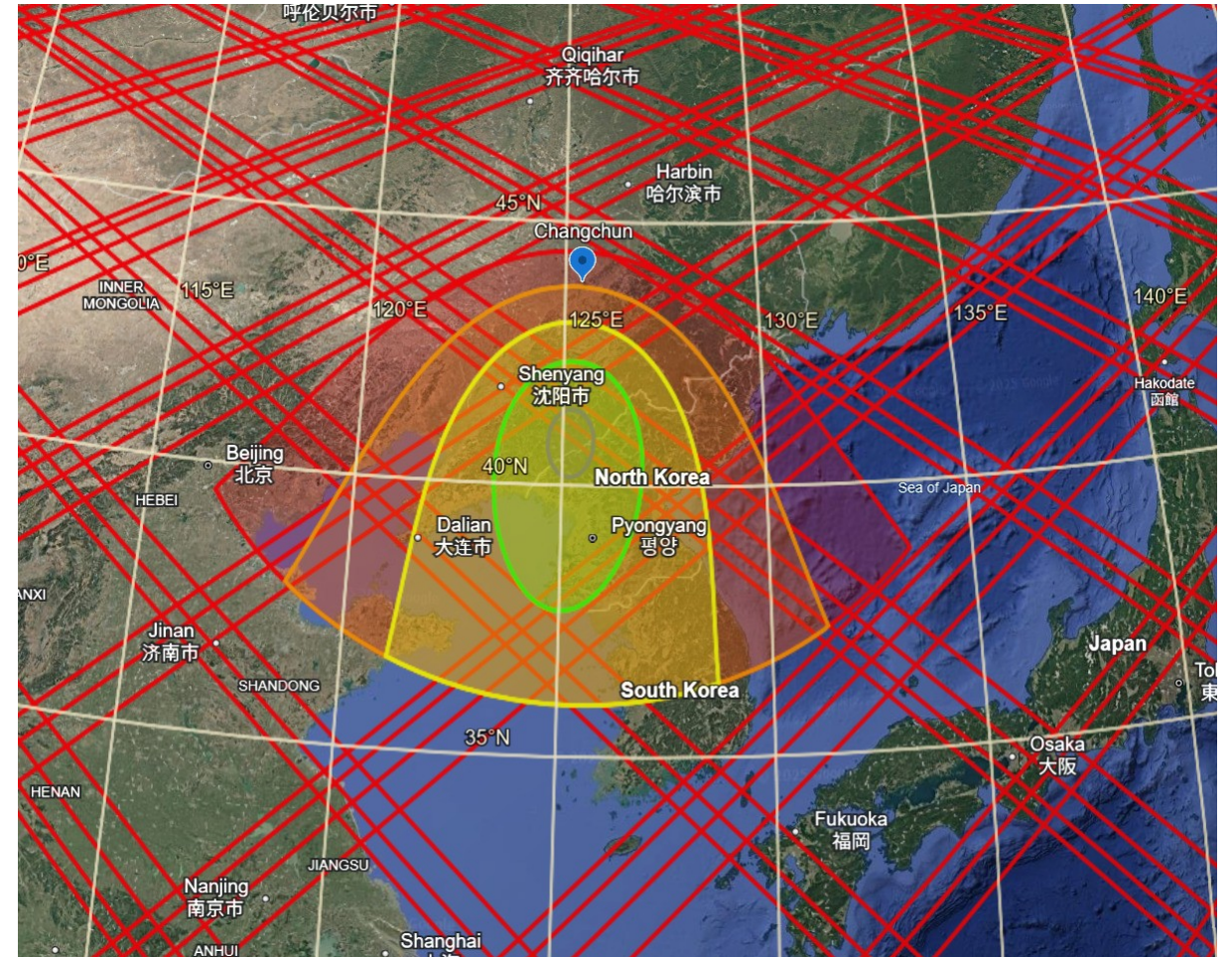
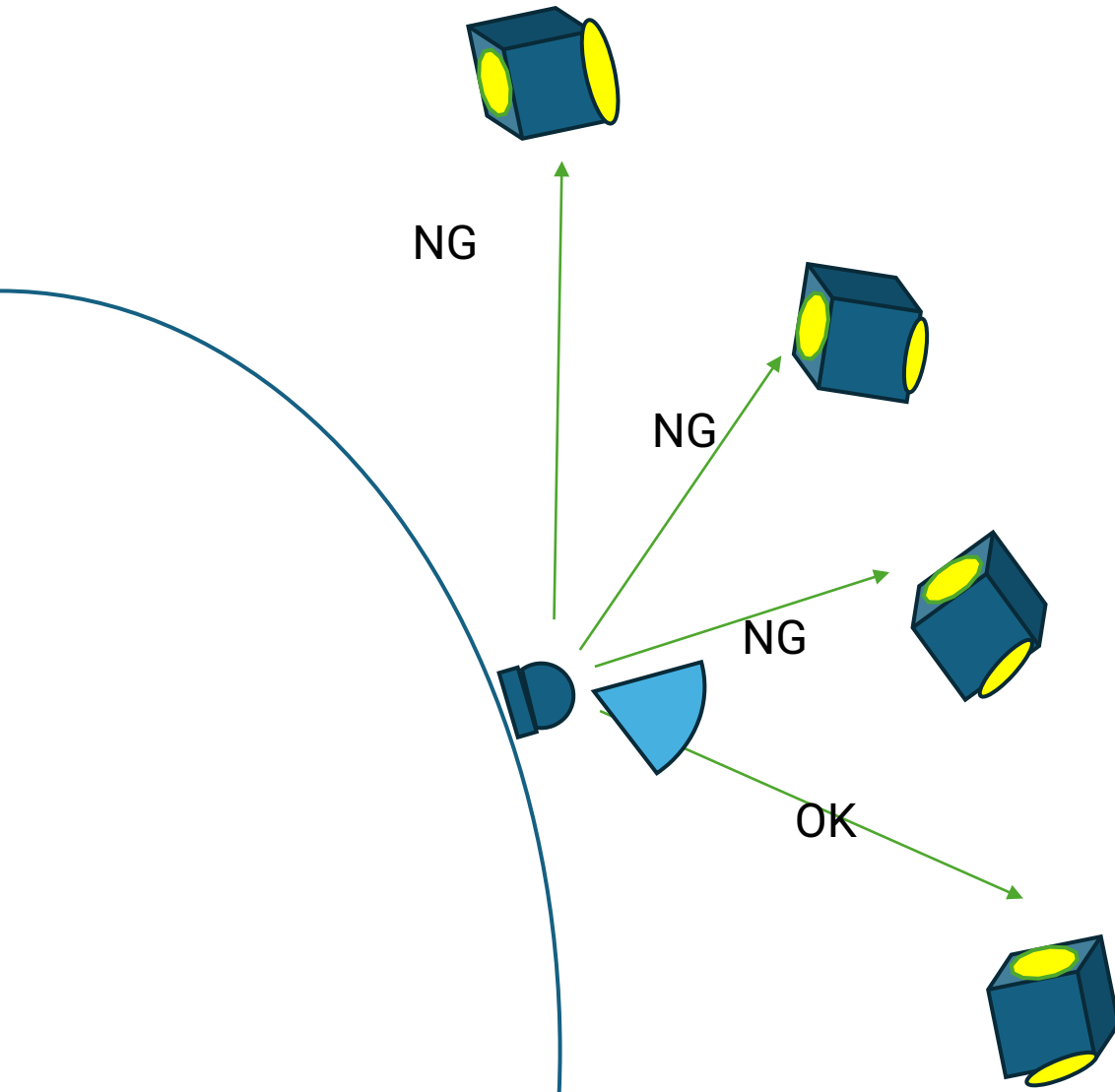
High-latitude stations (Latitude >51)

Northern hemisphere SLR station

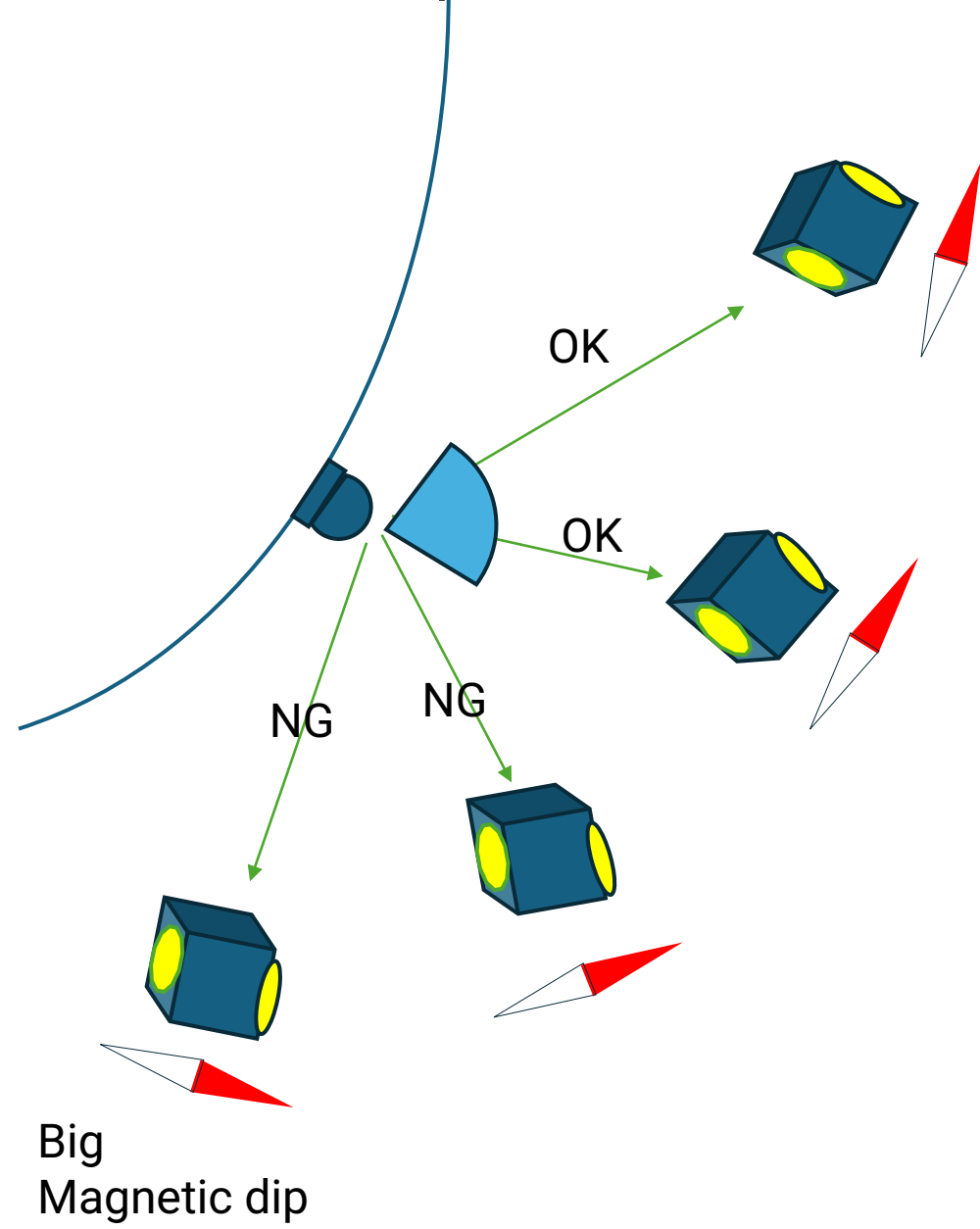


Middle latitude stations

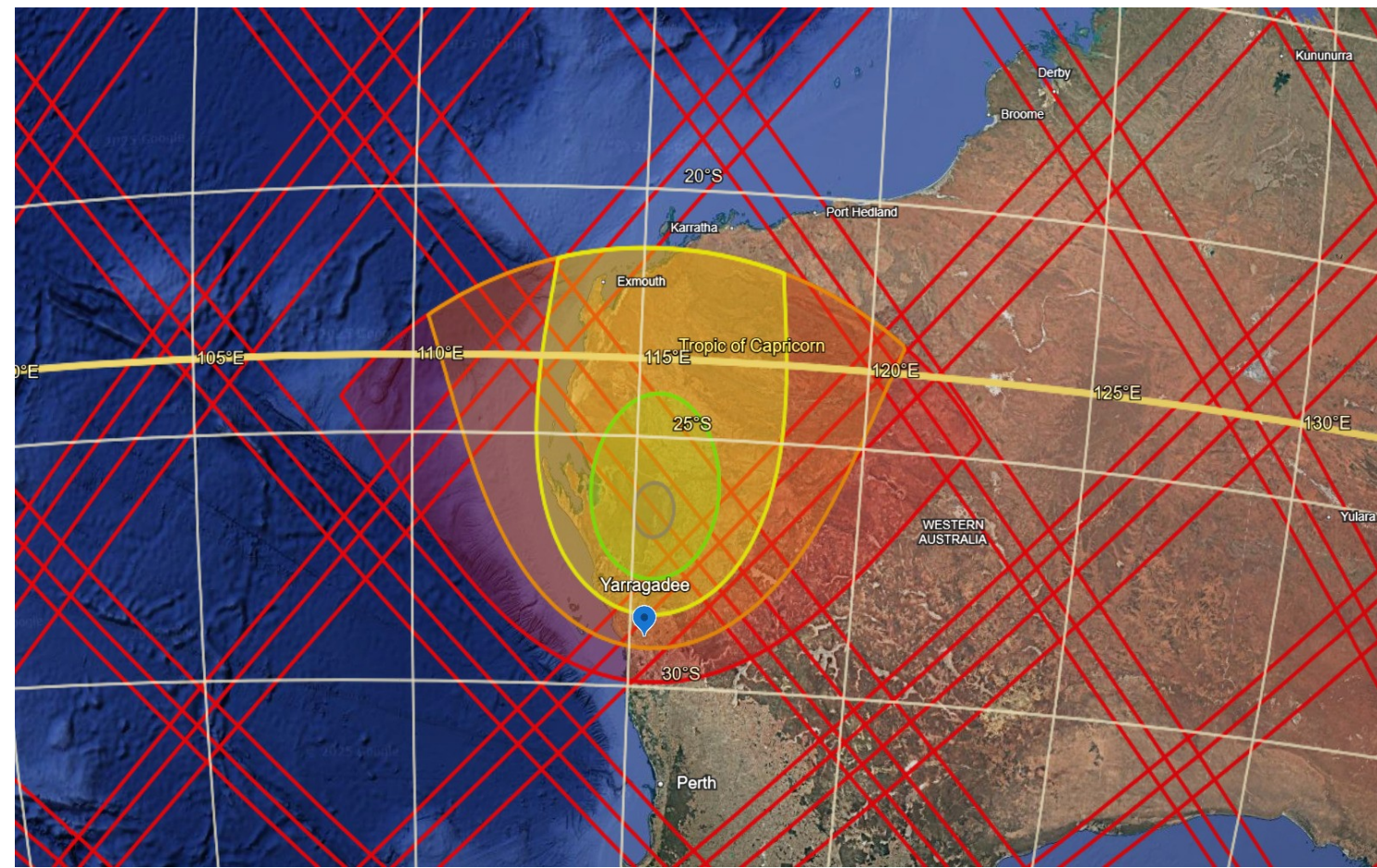
northern hemisphere SLR station



Southern hemisphere SLR station



Middle latitude stations



Restriction by NASA ISS team

NASA prohibits SLR observations for 10 days after a new spacecraft is launched toward the ISS or after it departs from the ISS.

The next SLR observation of the ekagaku1 satellite is on December 16.

Future Plan (Plan B)

Today – 2026/Jan End : JAXA will make CPF of TLE-65942 BOTAN

If No return

2026/Feb – Future : JAXA will make CPF of TLE-65941 YOTSUBA

Conclusion:

Even if the ekagaku1 satellite is visible, you will not be able to receive a return signal due to the relative positions of the SLR station and the ekagau1 satellite.

Northern Hemisphere SLR stations: You can observe the ekagaku1 satellite when it is located between your station's south to the zenith.

Southern Hemisphere SLR stations: You can observe the ekagaku1 satellite when it is located between your station's north to the zenith.

Currently, our guess is:

TLE-65941 Clover <-- TLE is correct

TLE-65942 BOTAN <-- TLE is misassigned. TLE gives name "ekagaku1" now.

TLE-65943 ekagaku1 <-- TLE is misassigned. TLE gives name "BOTAN" now.

If we receive an SLR response from TLE-65943, we plan to report the TLE misassignment to the SpaceTrack.org based on that evidence.

We apologize for putting you through this tedious observation, and we sincerely appreciate your cooperation.

Thank you Shinichi Nakamura

Recent discussion topics of the ASC connected to NESC

ILRS core sites, pass performance criteria, satellite priorities, and FR/NP data submissions

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Phone: 089 289 23735

ILRS NESC, 11.12.2025, online

ILRS core sites

The current list of ILRS core sites should be updated to account for major updates of some stations, etc. The existing list was slightly modified in 2023 last (cf. right plot).

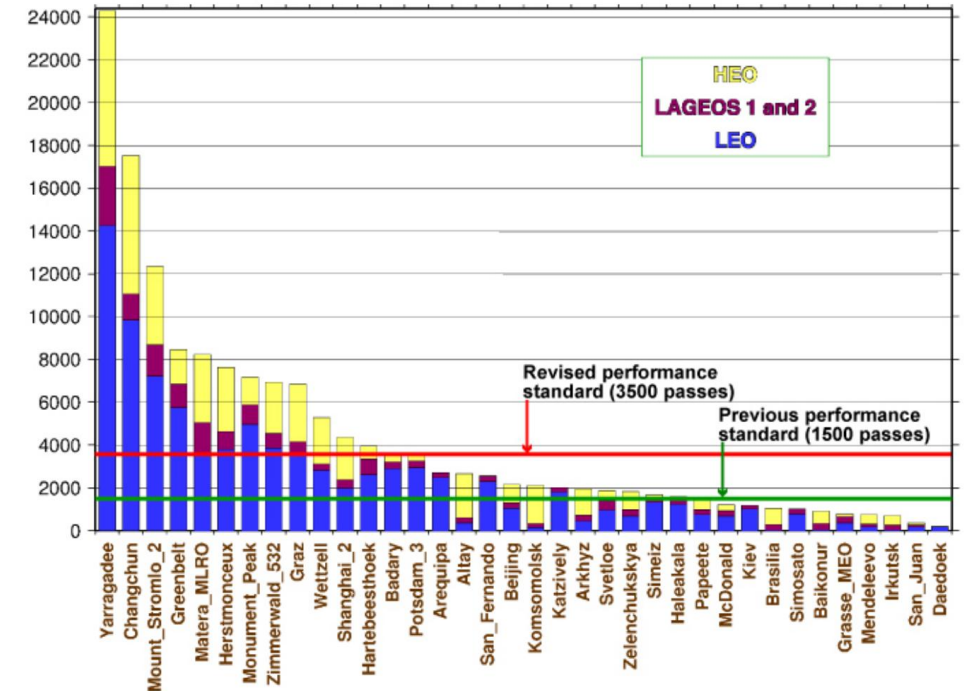
- The ASC initiated a task force to revisit the criteria (e.g., observation time span and coverage, discontinuities, stability and consistency of range biases). This task force should also
 - assess the current ILRS tracking network quality (performance/quantity, quality, etc.)
 - screen the combined weekly v280/v320 data to identify core stations
 - compile guidelines and a recommendation to GB (core site list + description how they had been identified)
 - Include information in new section in ILRS DHF
- Does anybody from NESC wants to join this task force?
(current members: L. Geisser, V. Husson, D. Sarrocco, M. Bloßfeld)

40442M006	7080	-7305.50000	17897.50000
50107M001	7090	-7670.50000	17897.50000
40451M105	7105	-6939.50000	17897.50000
40433M002	7109	-6939.50000	-731.50000
40497M001	7110	-6939.50000	3286.50000
40445M004	7119	4168.50000	17897.50000
40445M001	7210	-2160.50000	1825.50000
21611S001	7237	4382.50000	17897.50000
42202M003	7403	-3652.50000	365.50000
30302M003	7501	-0.50000	17897.50000
14001S007	7810	-730.50000	2960.50000
14001S001	7810	3022.50000	17897.50000
50119S003	7825	1460.50000	17897.50000
20101S001	7832	365.50000	17897.50000
14201S002	7834	-8766.50000	-2923.50000
10002S001	7835	-4109.50000	2190.50000
14106S009	7836	-2556.50000	1825.50000
21605S001	7837	-1095.50000	2190.50000
11001S002	7839	-6209.50000	17897.50000
13212S001	7840	-6209.50000	17897.50000
14106S011	7841	1491.50000	17897.50000
50119S001	7849	-730.50000	1459.50000
42202S001	7907	-8766.50000	-2558.50000
12734S001	7939	-6209.50000	364.50000
12734S008	7941	365.50000	17897.50000
14201S018	8834	-1340.50000	3336.50000

ILRS station pass performance criteria

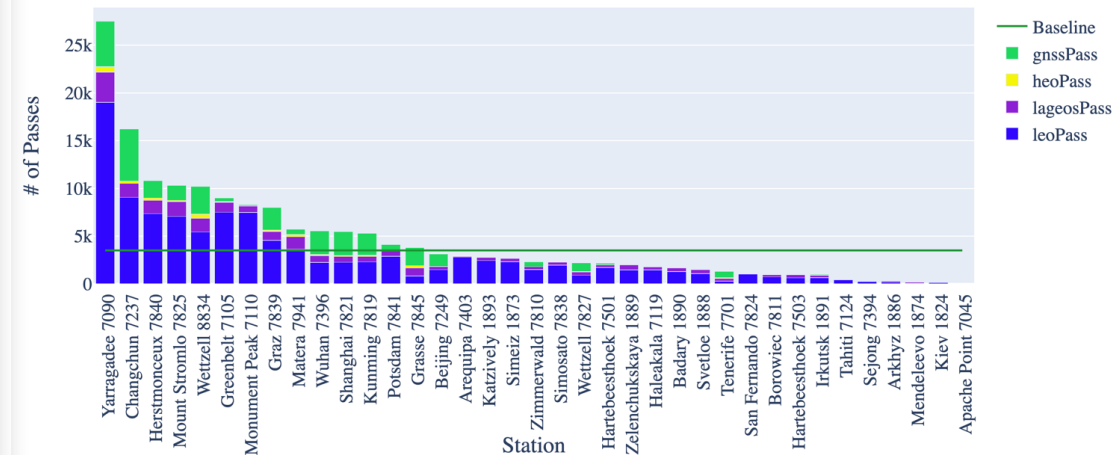
- More and more satellites are in orbit which require SLR tracking!
- ILRS GB raised pass performance standards from 1500 passes (1996) to 3500 passes (2015)
- BUT: only 25% of stations reach these standards, i.e. contribute significantly to the TRF!
- Now, a raise to 4900 passes will be proposed to the GB
 - kHz laser systems, new generation of event timers, new stations, ...
- What is the opinion of the NESC here?
- The ASC is thinking to assess the need how many passes are needed to obtain reliable/stable station coordinates (via simulations).

total passes
from October 1, 2014 through September 30, 2015



20151003

Total Number of Passes (All Satellites)
(2022-01-01 to 2022-12-31)



ILRS satellite priorities

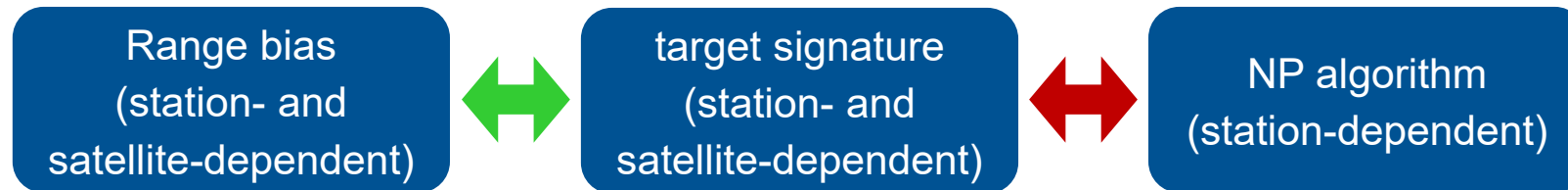
The ILRS has ordered its tracking priority list according to [satellite orbital parameters](#) and [special project needs](#).

- Priorities decrease with:
 - increasing orbital altitude, and
 - increasing orbital inclination (at a given altitude)
- Priority of some satellites may then be increased to intensify support for:
 - active missions (such as altimetry), special campaigns, or
 - post-launch intensive tracking phases, and
 - some slight reordering may be done to give higher priority to missions with increased importance to the analysis community.
- Stations may adjust priorities to accommodate [local conditions](#) such as system capabilities, weather, and special program interests.
- Unfortunately, very important satellites like LA-1/-2 are at place 24/25 → creates confusion, maybe also leads to less LA-1/-2 passes per station!
- The ASC discussed at its last meeting if one should define the priority list not based on height but category (e.g., high priority of ITRF satellites). What's the NESC opinion on this?

Priority	Mission	ILRS Name	COSPAR ID	SIC	Sponsor	Altitude (km)	Inclination (degrees)	Comments
1	GRACE-FO-1/2	gracefo1 gracefo2	1804701 1804702	0123 0124	NASA JPL and the German Research Centre for Geosciences (GFZ)	500	89	1-month campaign
2	ICESat-2	icesat2	1807001	6873	NASA	496	92	Restricted tracking; authorization required
3	CryoSat-2	cryosat2	1001301	8006	ESA	450-720	92	
4	PAZ	paz	1802001	2501	HISDESAT	514	97.44	
5	TanDEM-X	tandemx	1003001	6202	Infoterra/ DLR/ GFZ/CSR	514	98	Tandem with TerraSAR-X
6	TerraSAR-X	terrasarx	0702601	6201	Infoterra/ DLR/ GFZ/CSR	514	97.44	Tandem with TanDEM-X
7	Sentinel-3B	sentinel3b	1803901	8011	ESA/ EUMETSAT	814.5	98.65	Restricted tracking; authorization required
8	Sentinel-3A	sentinel3a	1601101	8010	ESA/ EUMETSAT	814.5	98.65	Restricted tracking; authorization required
9	SWOT	swot	2217301	4381	NASA/CNES	857 then 890	77.6	
10	Swarm-A/C	swarma swarmc	1306702 1306703	8007 8009	ESA	460 530	88.35 88.95	
11	Swarm-B	swarmb	1306701	8008	ESA	460	88.35	
12	SARAL	saral	1300901	3201	CNES and ISRO	814	98.55	
13	HY-2D	hy2d	2104301	2211	China	971	66	
14	HY-2C	hy2c	2006601	2209	China	957	66	
15	HY-2B	hy2b	1808101	2208	China	971	98.55	
16	Sentinel-6A/ Jason-CSA	sentinel6a	2008601	4380	NASA, ESA, EUMETSAT, NOAA, CNES	1339.4-1355.9	66.042	
17	Jason-3	jason3	1600201	4379	NASA, CNES, Eumetsat, NOAA	1,336	66.0	
18	LARES-2	lares2	2208001	5988	ASI	5,899	70.16	
19	Larets	larets	0304206	5557	IPIE	691	98.204	
20	Starlette	starlette	7501001	1134	CNES	815-1,100	49.8	
21	Stella	stella	9306102	0643	CNES	815	98.6	
22	WESTPAC	westpac	9804305	8801	Western Pacific Laser Tracking Network	835	98	Resumed tracking on 2025-03-17
23	LARES	lares	1200601	5987	ASI	1,450	69.5	
24	LAGEOS-2	lageos2	9207002	5986	ASI/NASA	5,625	52.6	
25	LAGEOS-1	lageos1	7603901	1155	NASA	5,850	109.8	
26	Etalon-1	etalon1	8900103	0525	Russia	19,100	65.3	
27	Etalon-2	etalon2	8903903	4146	Russia	19,100	65.2	
28	Ajisai	ajisai	8606101	1500	JAXA	1,485	50	
29	IRNSS-1J	irnss1j	2307601	3310	ISRO	35,790	5	
30	GLONASS-141	glonass141	1908801	9141	Russia	19,140	65	
31	Galileo-102	galileo102	1106002	7102	ESA	23,220	56	

ILRS FR/NP data submission

Within the analysis of SLR observations, the **range bias** is highly correlated with the **satellite target signature**, which itself depends partly on the applied **algorithm to compute a NP** at each station



- Based on a former request (NESC/ASC?), the ILRS network stations should upload the computed NPs as well as the respective FR data to the ILRS data portals (format: CRDv2)
- Up to now, stations upload “only” the clipped FR data used to generate the NP to EDC/CDDIS
- With this data set, we cannot recompute the NP using other algorithms which means we cannot assess the impact of the NP algorithm on the target signature model (red arrow)
- The ASC wants to approach the stations (via CB) to stress the need for publishing the unclipped FR data again (with the help of NESC?)
- Afterwards, an ASC task force might investigate the impact of a commonly applied NP algorithm for analysis products (and models)

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Academy of Sciences



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